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FRATTURA  
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Torino, November 5, 2008

# **Fatigue Crack Initiation, Propagation and Failure at Very-High Numbers of Cycles**

**Stefanie E. Stanzl-Tschegg**

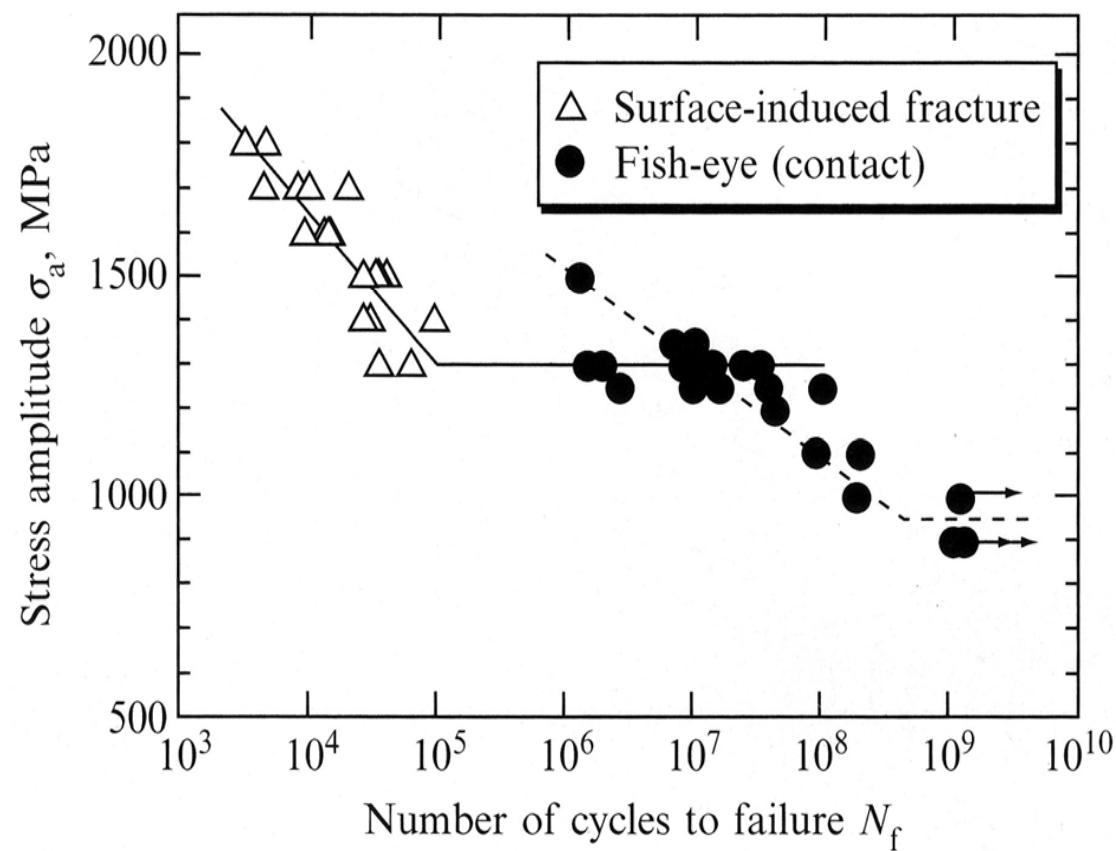


**BOKU University of Natural Resources and Applied Life Sciences  
Vienna, Austria**

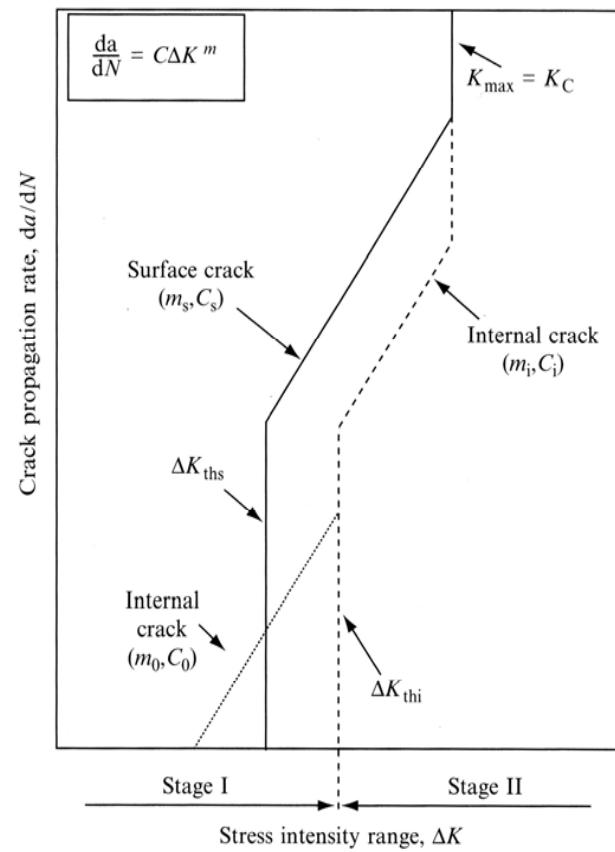
# Contents

- **Very high cycle fatigue (VHCF), fatigue crack growth (FCG) and  $\Delta K$  thresholds at ultrasonic frequency**
  - Experimental procedure and results:
    - S-N and  $(\Delta a/\Delta N - \Delta K)$  curves of **different materials**
    - Influence of environment and frequency
- **Plastic deformation, crack initiation, propagation and failure in copper in the VHCF range (20 kHz)**
  - Persistent slip bands (PSBs)
  - Endurance limit ,  $\Delta a/\Delta N$  and  $\Delta K$  threshold

# VHCF and FCG Behaviour

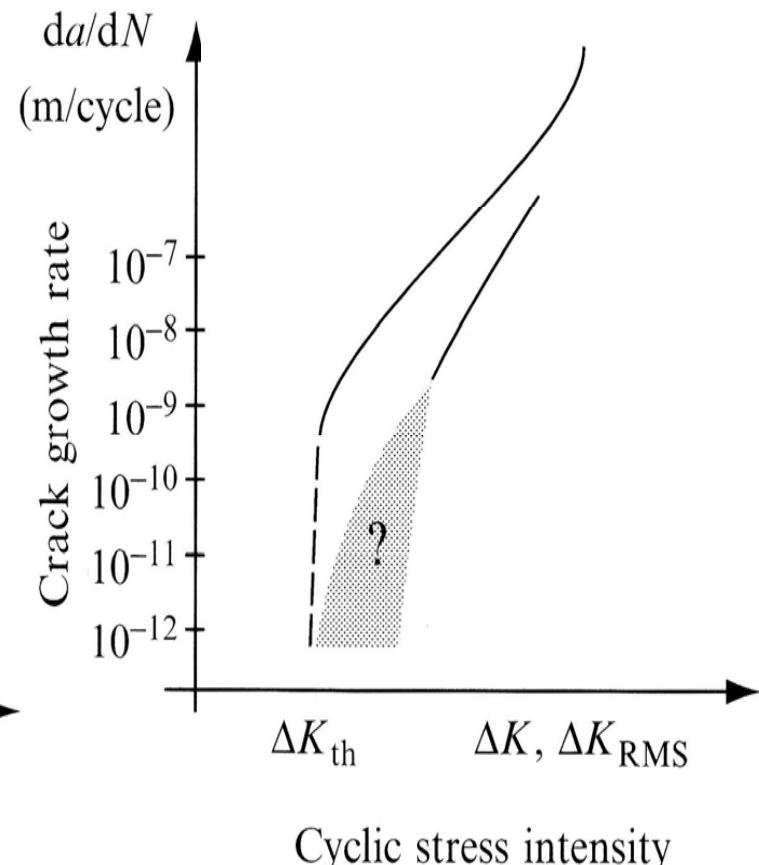
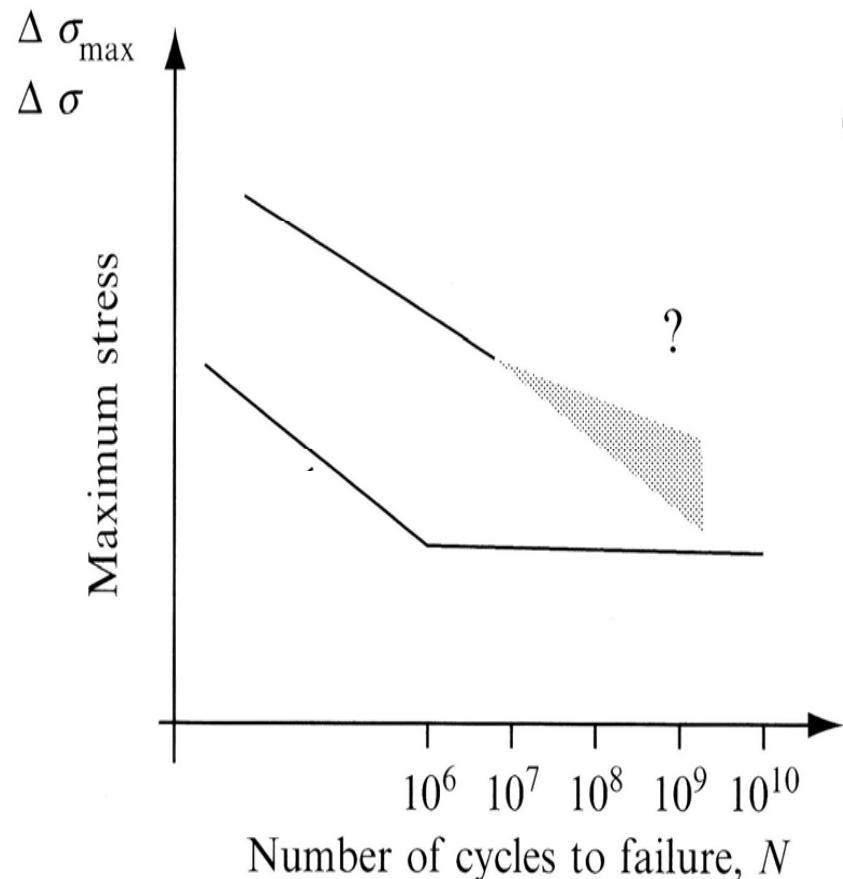


S-N diagram of high-carbon  
Cr-steel (Sakai et al 2002)



FCG behaviour of  
Surface and internal cracks  
(Tanaka & Akiniwa 2002)

# Existence or Non-Existence of Endurance Limit and Threshold Stress Intensity



# VHCF – Testing Times

Number of cycles	20 Hz	20 kHz
$10^6$	14 hours	50 sec
$10^7$	6 days	8 min
$10^8$	2 months	1.4 hours
$10^9$	1.7 years	14 hours
$10^{10}$	17 years	6 days
$10^{11}$	170 years	2 months



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BOKU University  
1915

# Ultrasound Measuring Technique



BOKU University  
2004

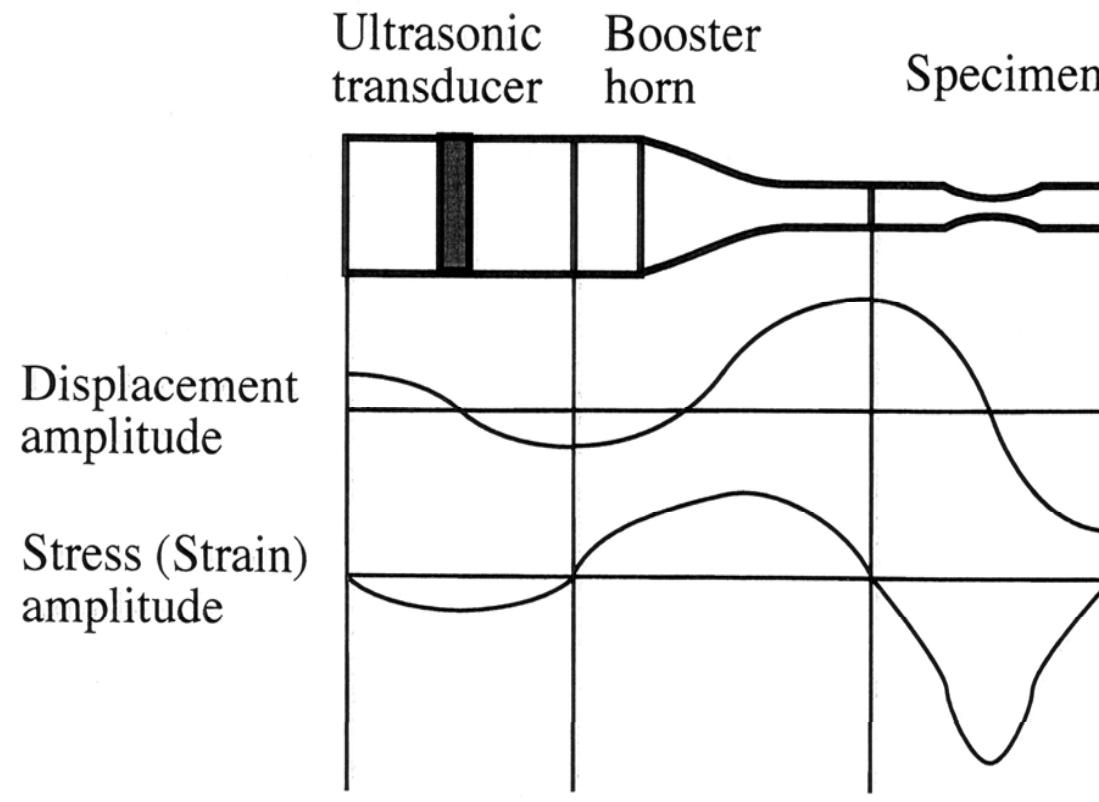


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# Principle of Ultrasonic Resonance Fatigue Loading



Resonance vibration at approximately 20 kHz

- **Ultrasonic Transducer:** Piezoelectric, axial or torsional
- **Magnification Horn:** Increases the vibration amplitude by a factor of 4 to 15
- **Specimen:** Length and shape follows resonance criteria,  
Surface condition is important  
Smooth or notched



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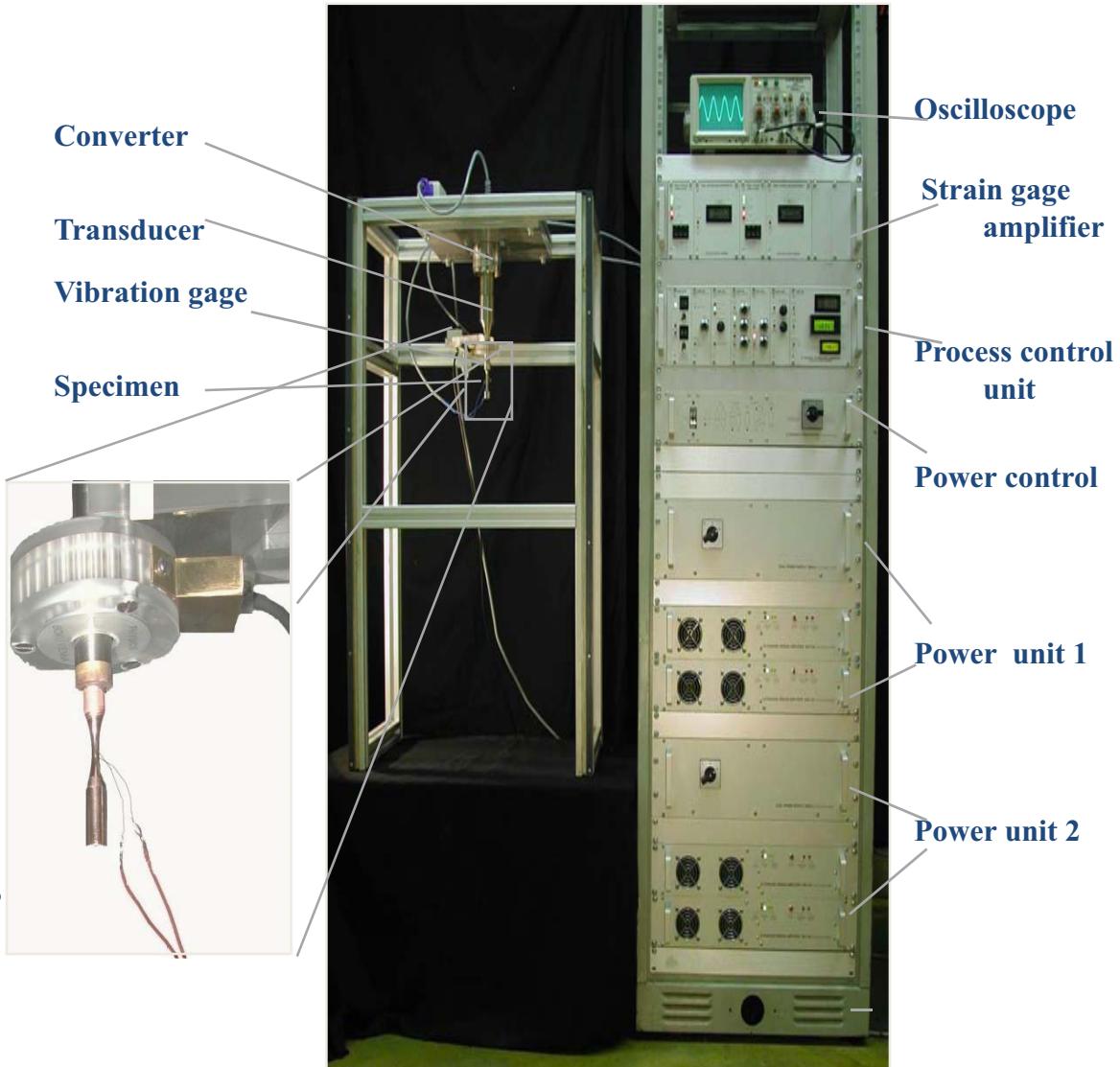
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# Ultrasonic Fatigue Loading – Experimental Procedure

- Frequency 19 kHz
- Fully reversed loading ( $R=-1$ )
- Cooling (damping heat): compressed air pulse-pause sequences

pulses: 150 ms (2700 cycles)  
periodic pauses: 2000-3000 ms
- Induction sensor measures vibration amplitudes
- Closed loop control of vibration amplitudes (total strain)

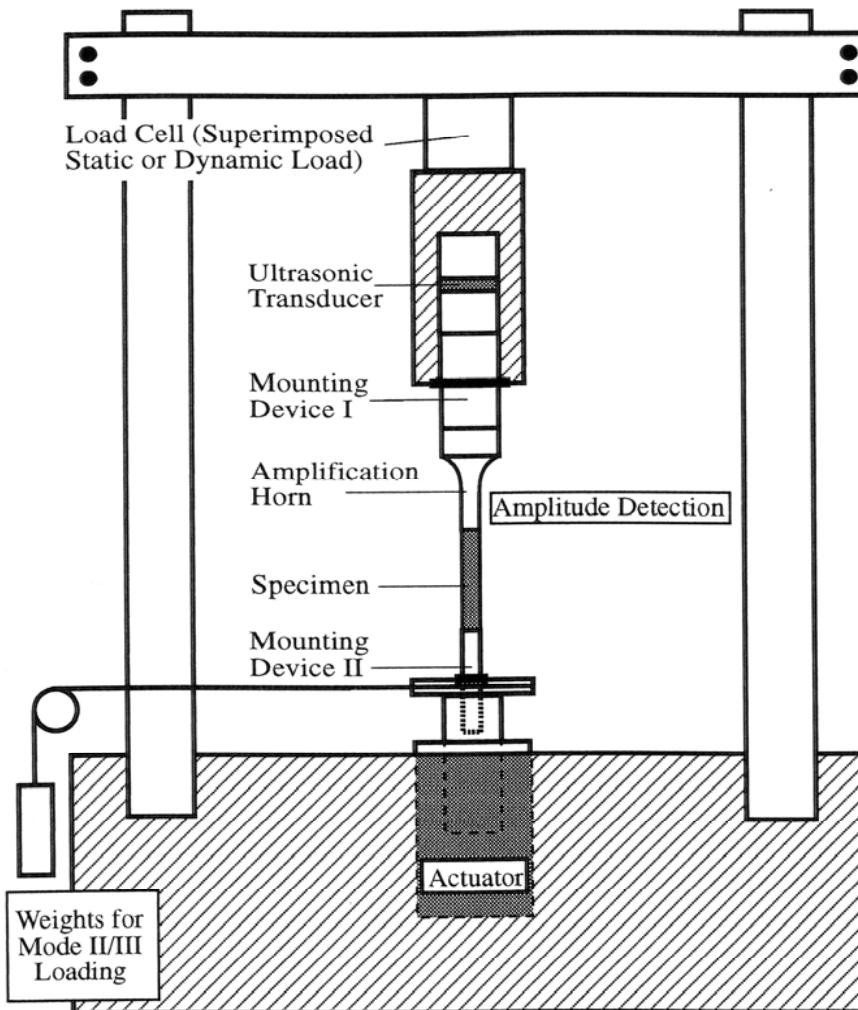


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# Ultrasound Testing Equipment



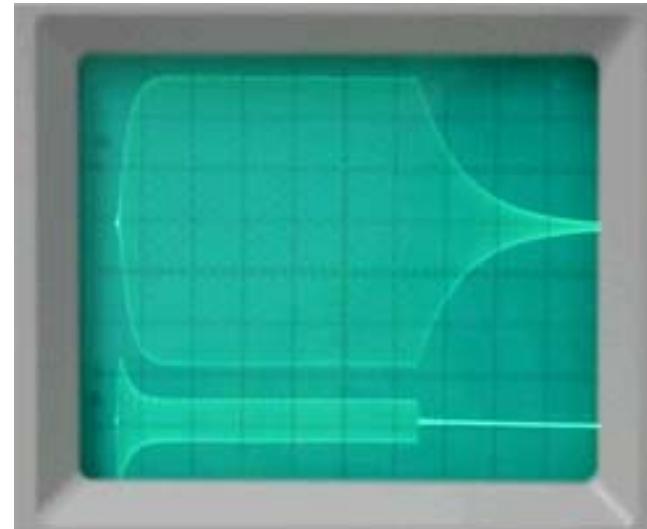
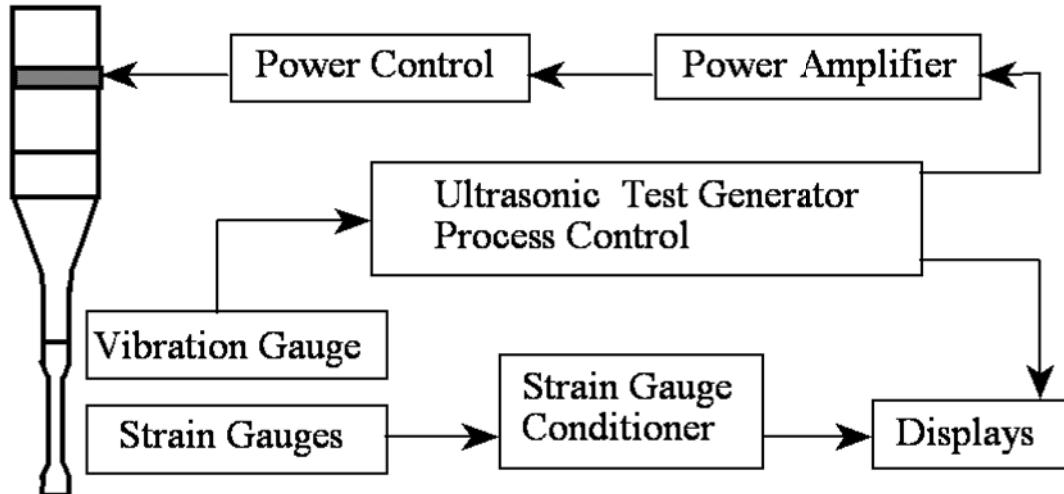
- Amplitude sensor

$$\Delta I \rightarrow \varepsilon = \Delta I / I \rightarrow \\ \sigma = E \cdot \varepsilon \rightarrow \Delta K$$

- Free specimen end  $\rightarrow R = -1$
- Both specimen ends fix  $\rightarrow R \geq 1$
- Cooling Pulse-Pause  
+ Compressed air
- Environmental chambers  
(corrosion, temperature, vacuum)

# Ultrasonic Testing Procedure

## Amplitude feed-back control



### Determination of the process:

Vibration Amplitude: After calibration

Pulse and Pause: Depending on damping

Frequency limits: Stop when specimen failure

### Control Circuits:

Vibration Amplitude: Accuracy  $\pm 1\%$

Resonance Frequency: Accuracy  $\pm 1 \text{ Hz}$  ( $0.005\%$ )

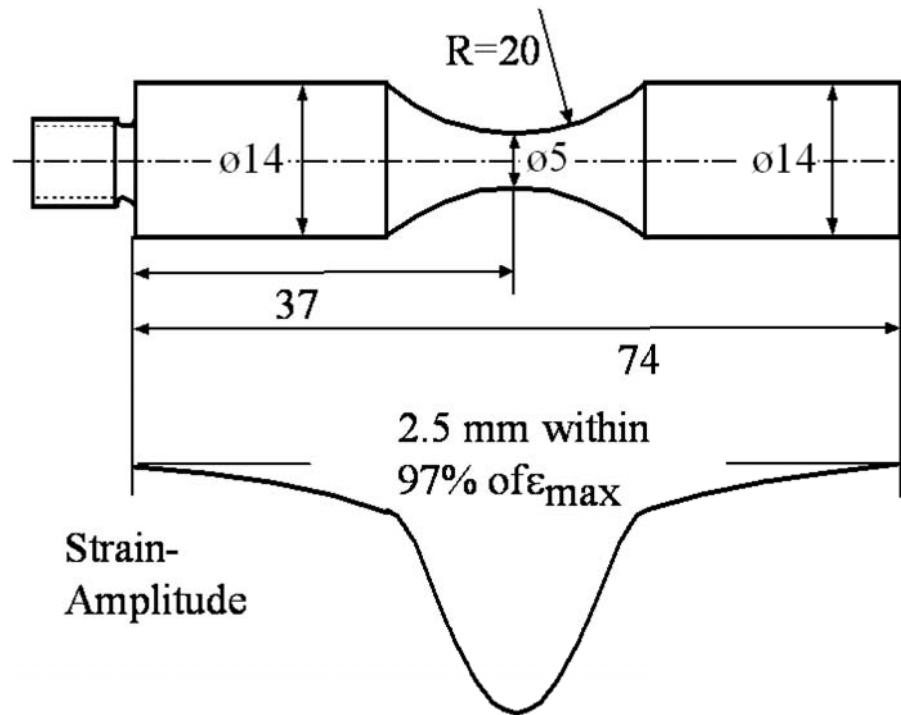
Pulse 50 ms - approx. 1000 amplitudes

Increase Time - approx. 100 amplitudes

Decay Time - depending on damping

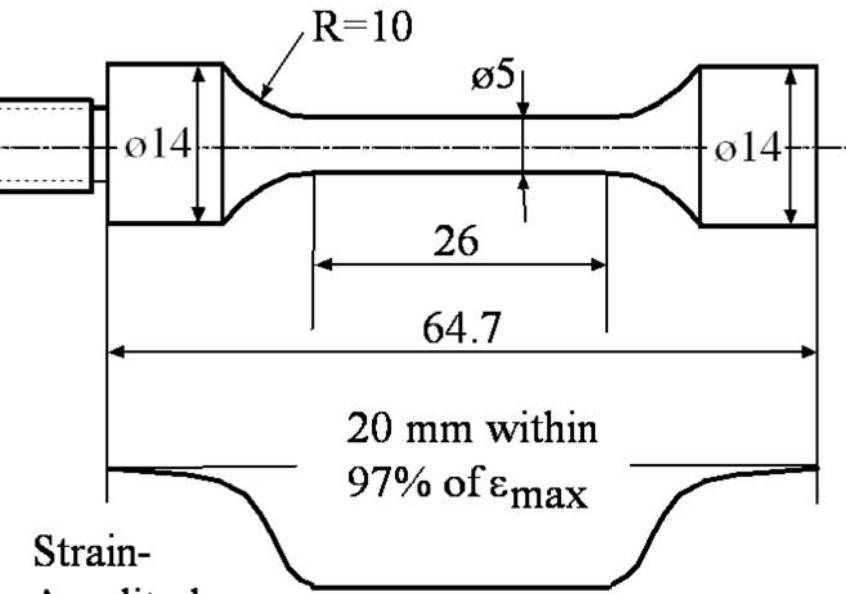
# Shapes of Fatigue Specimens

Wrought alloy



Amplification factor: 4.9  
Loaded volume:  $50 \text{ mm}^3$

Cast alloy



Amplification factor: 2.2  
Loaded volume:  $390 \text{ mm}^3$

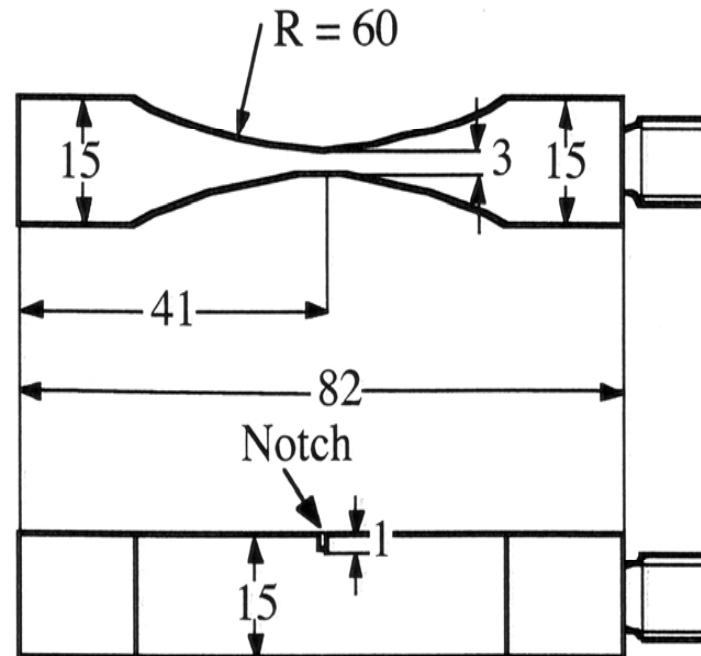


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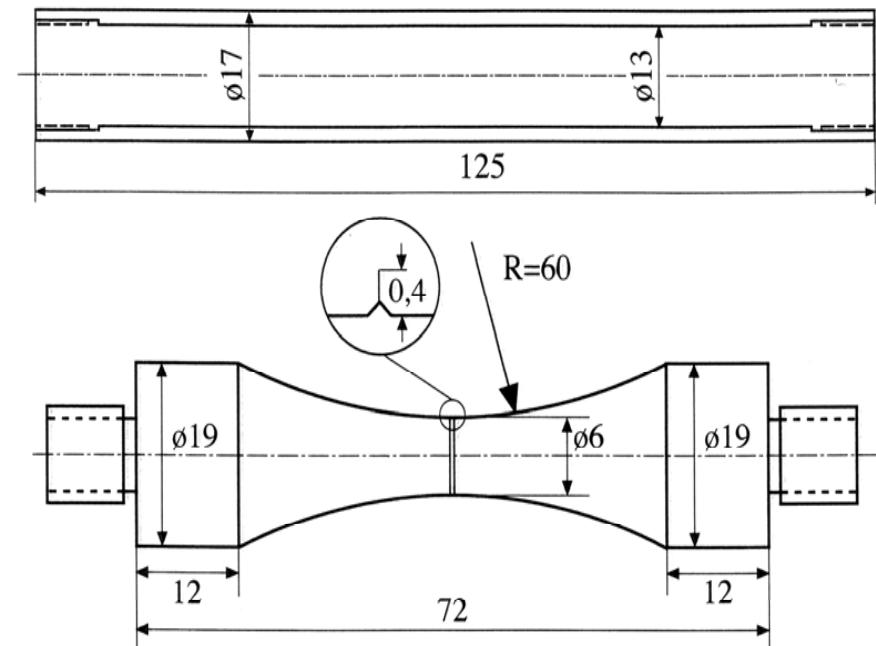
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# Specimen Shapes - Fatigue Crack Growth



Fully Reversed Loading



Superimposed and Mixed  
Mode Loading

# Ultrasonic Fatigue Testing Procedure – Mechanical Components

## Testing in different environments

Vacuum

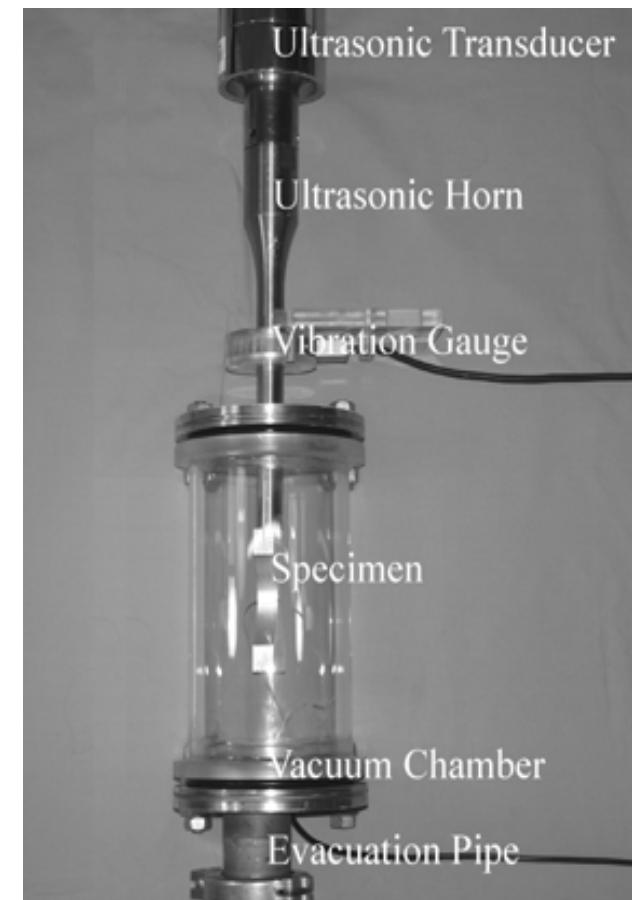
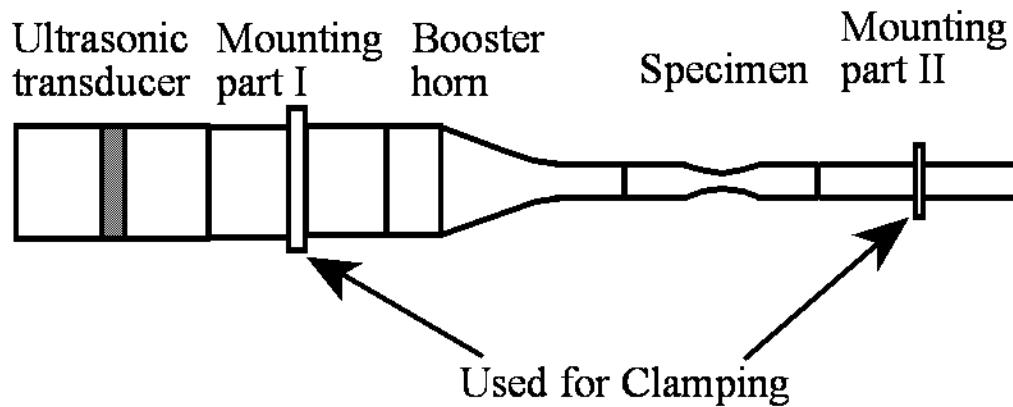
High and low temperature

Corrosive Fluids

## Testing at different loading conditions

Superimposed mean loads ( $R>-1$ )

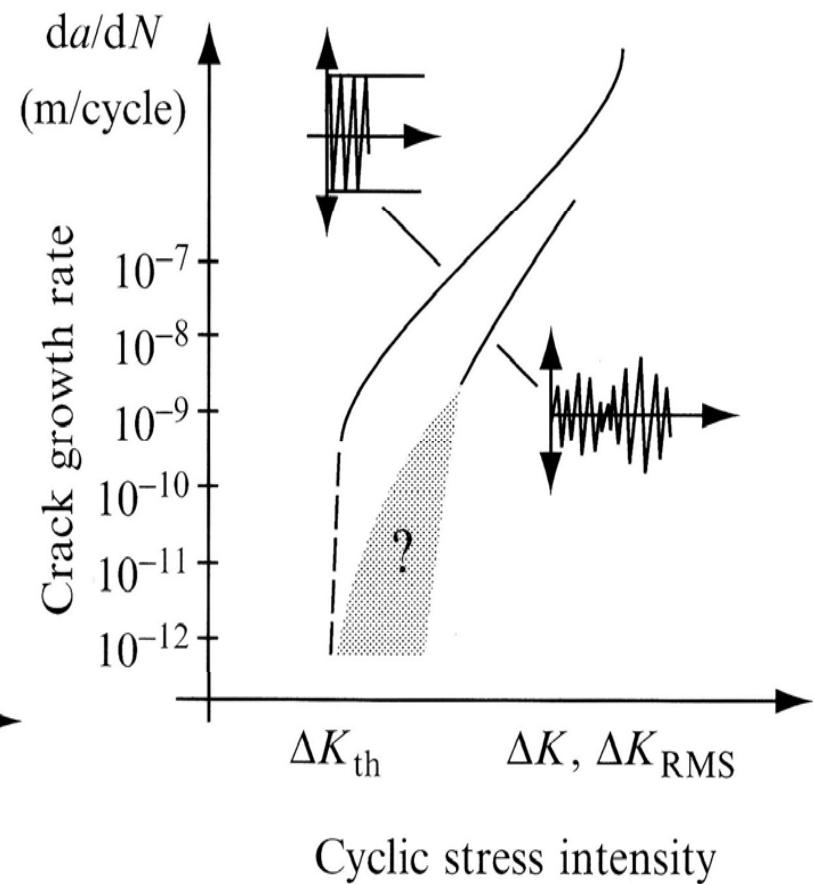
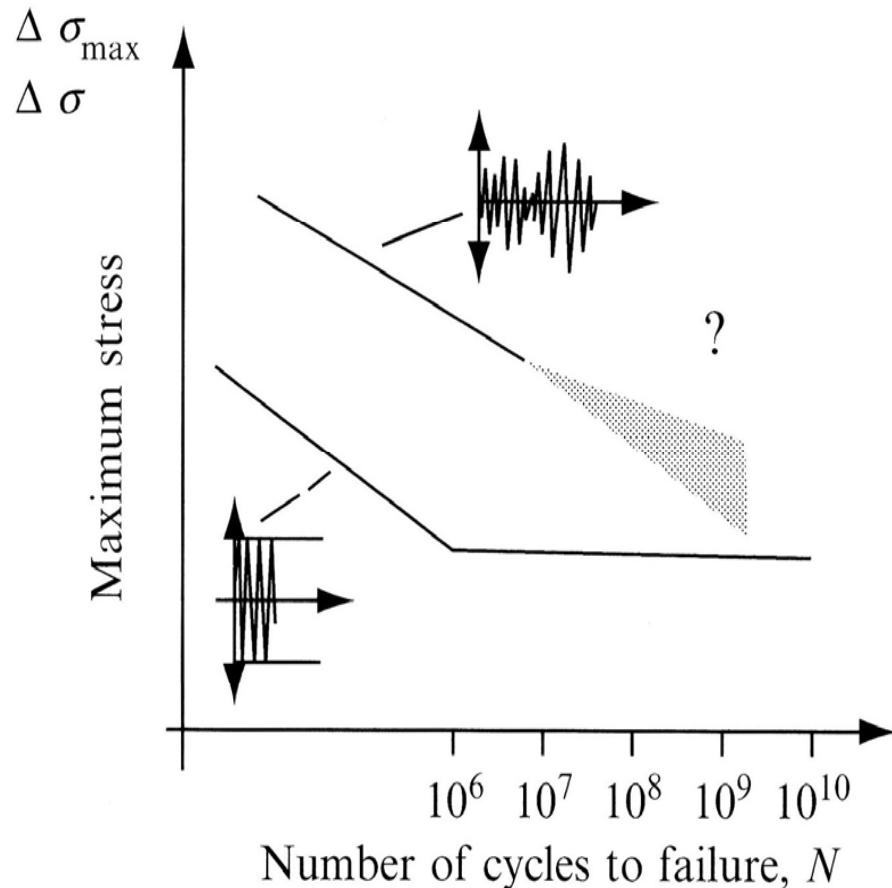
Superimposed torsion moments



*Testing in vacuum*

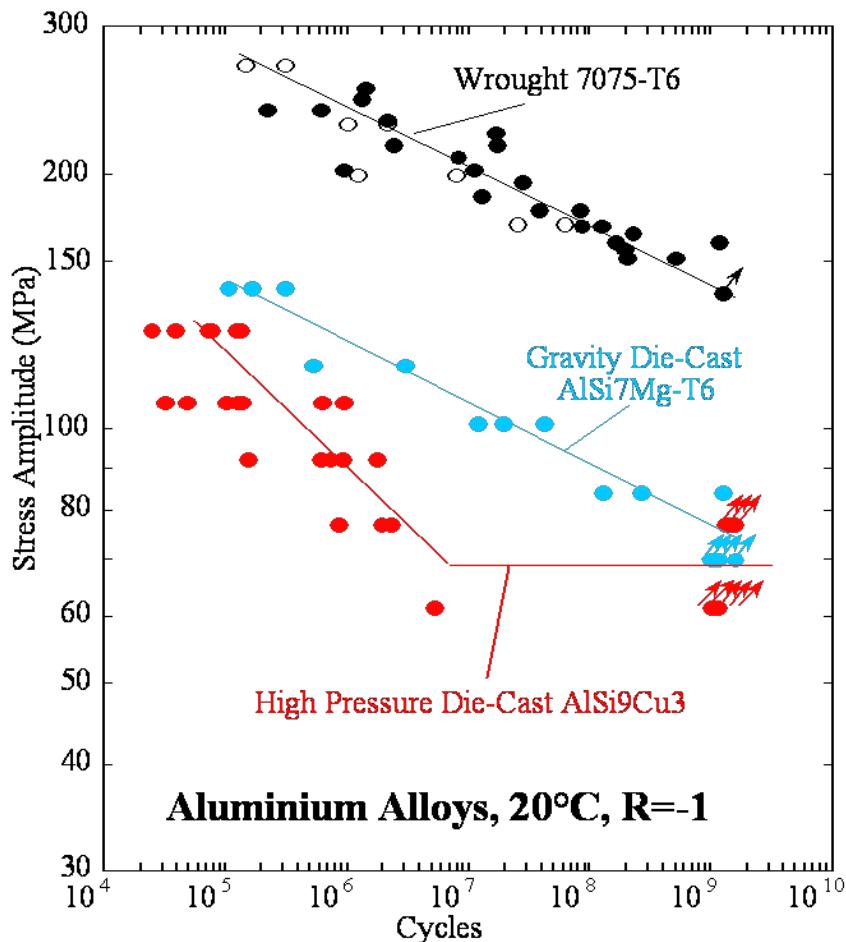
# Random loading

## Endurance Limit - Threshold Stress Intensity



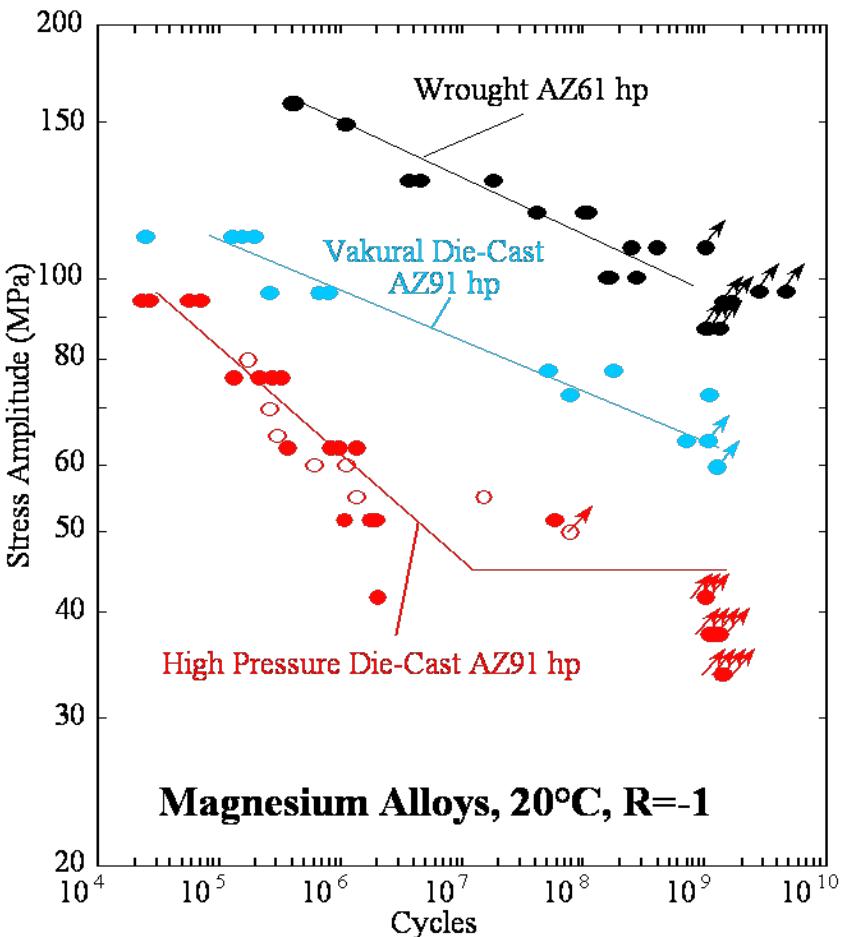
# VHCF Aluminium and Magnesium Alloys

## S-N Curves



Papakyriacou M, Mayer H, Fuchs U,  
Stanzl-Tschegg SE, FFEMS 25, 2002.

Stanzl-Tschegg SE, Proc.  
ATEM'93, 2003, 03-207



Mayer H, Papakyriacou M, Zettl B, Stanzl-  
Tschegg SE, Int J Fatigue, 25, 2003, 245-256

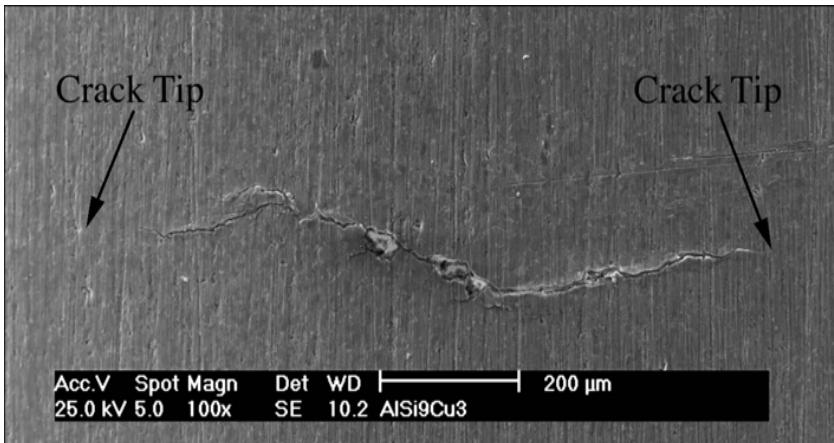


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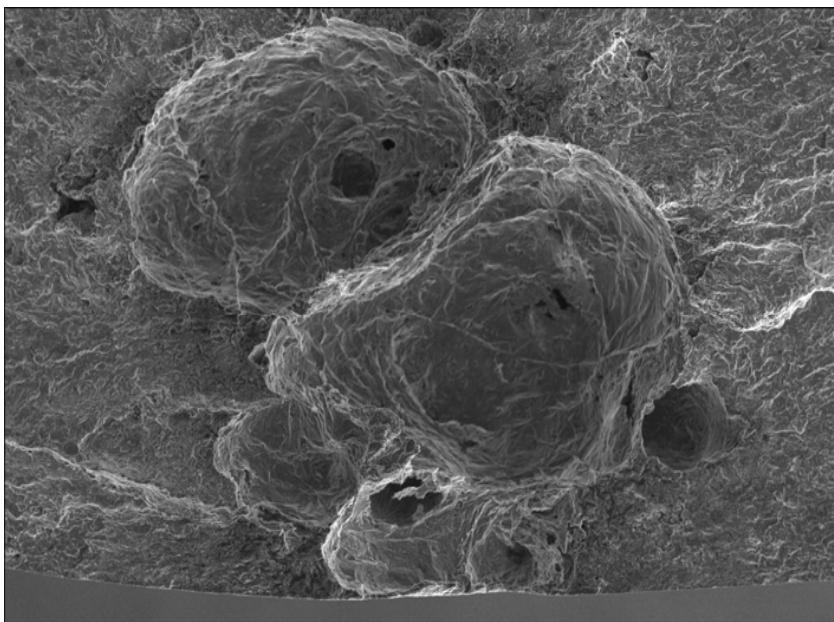
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# VHCF Aluminium and Magnesium Alloys



**Non-propagating crack**  
**High-pressure die cast AlSi9Cu3**

1.05x10<sup>9</sup> cycles, 47.5 MPa  
(78 % of mean endurance strength at 150 °C)



Maximum length of non-propagating cracks: 1.2 mm

**Fracture surface**  
(after increase of load)

Mayer H, Kolloquium Werkstoffe, Fertigung, Konstruktion, Siegen 2004

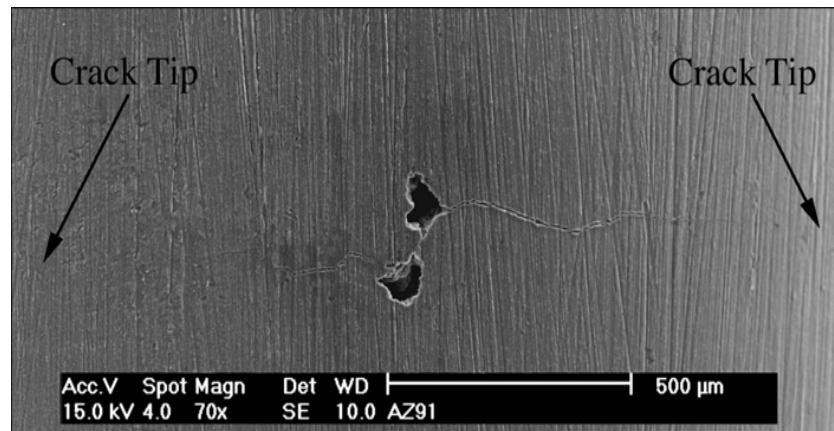


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# VHCF Aluminium and Magnesium Alloys



**Non-propagating crack  
High-pressure die cast AZ91**

$1.51 \times 10^9$  cycles, 38 MPa  
(90 % of mean endurance  
strength at 150 °C)

Maximum length of  
non-propagating crack: 1.6 mm

Fracture surface  
(after increase of load)

Mayer H, Kolloquium Werkstoffe, Fertigung, Konstruktion, Siegen 2004



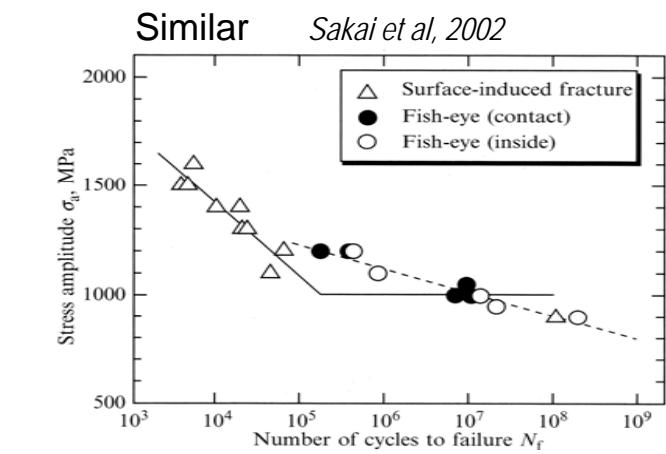
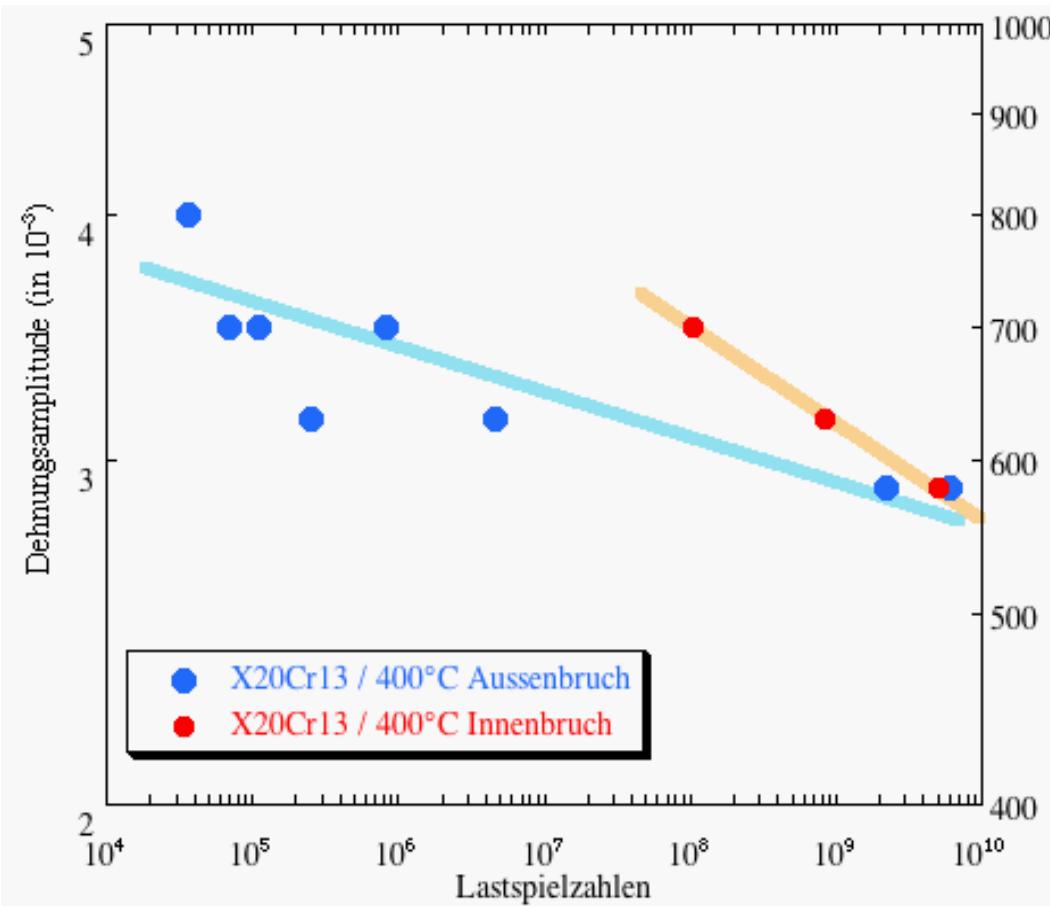
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# VHCF X20Cr13 Steel

Axial 20 kHz tension-compression loading  
X20Cr13: 11.3 Cr, 0.2 C, 0.22 Al  
(940°C /oil quench/ hardening 400°C/2.5 h)



- Crack initiation from surface
- Crack initiation  $\geq 10^8$  cycles from internal inclusions



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Stanzl-Tschegg, DGM Meeting Berlin, June 2006

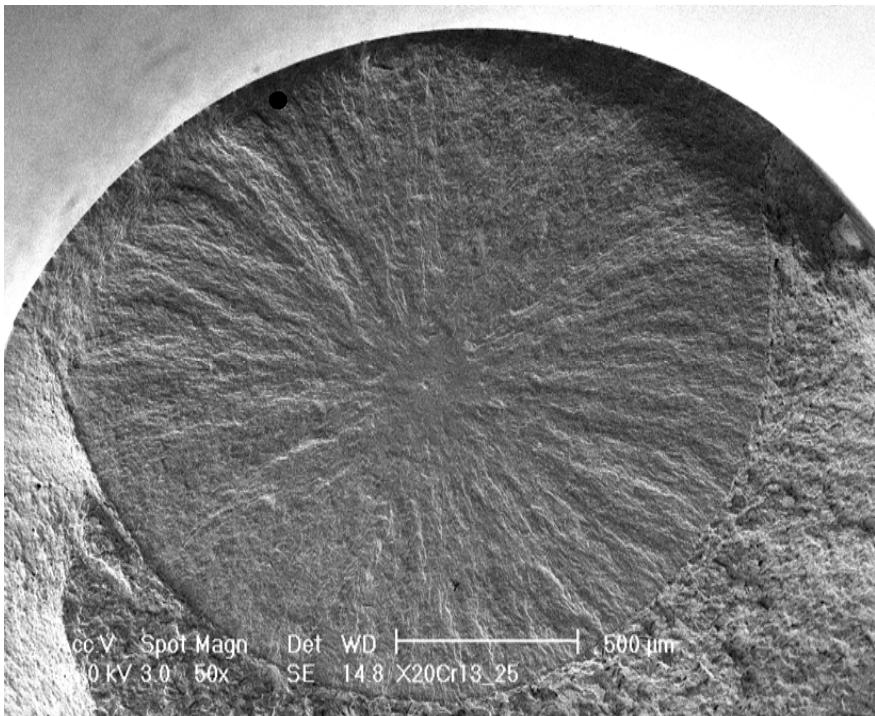
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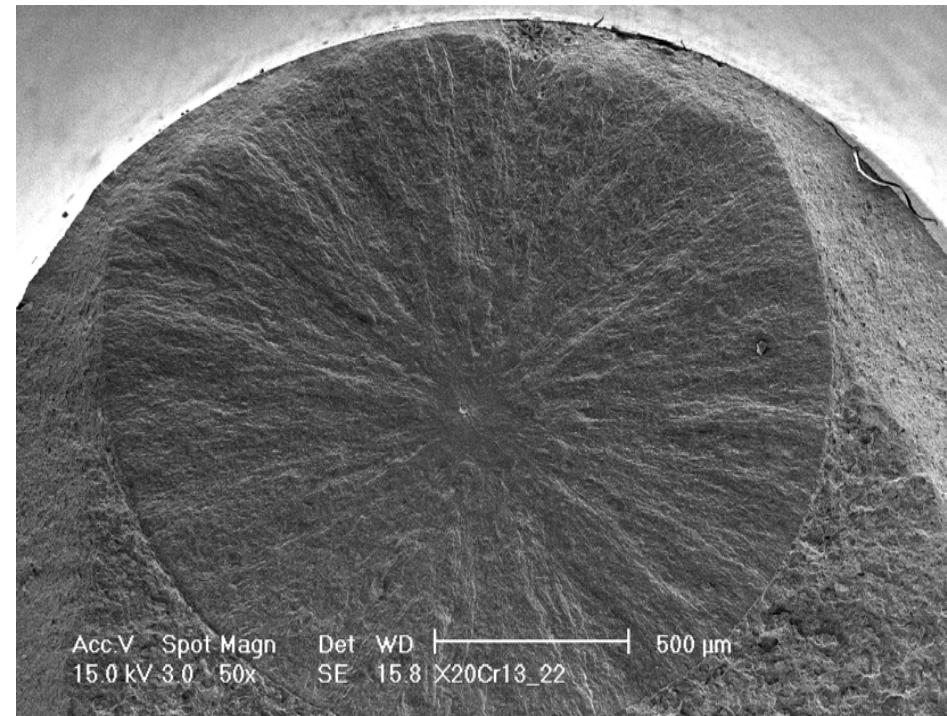
# Crack Initiation from Internal Inclusions

Axial 20 kHz Tension-Compression Loading

X20Cr13: 11.3 Cr, 0.2 C, 0.22 Al (940°C / oil quenched, hardening 400°C/2.5 h)



$\Delta\sigma = 620 \text{ MPa}$ ,  $N_f = 8.48 \times 10^8$



$\Delta\sigma = 700 \text{ MPa}$ ,  $N_f = 1.3 \times 10^8$

Stanzl-Tschegg, DGM Meeting Berlin, June 2006



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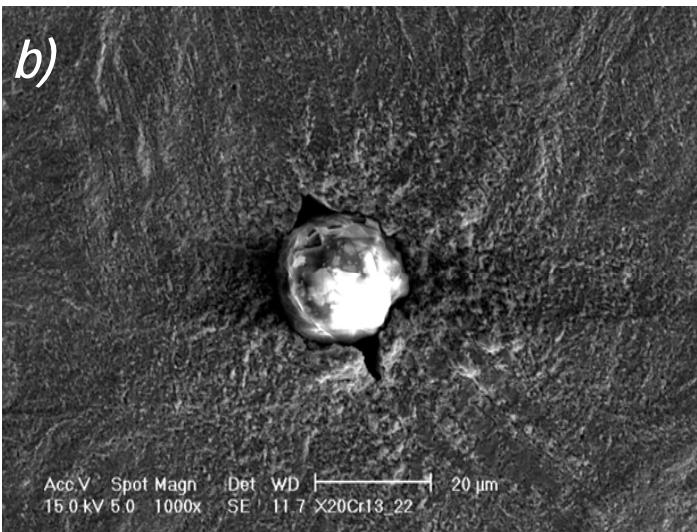
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# Crack Initiation from Interior Inclusions



Crack initiation from ca. 10-20  $\mu\text{m}$   
 $\text{Al}_2\text{O}_3$ , CaO, MnS, MgO inclusions  
after ca.  $10^8 - 10^9$  cycles



EDAX Analysis (Wt%)

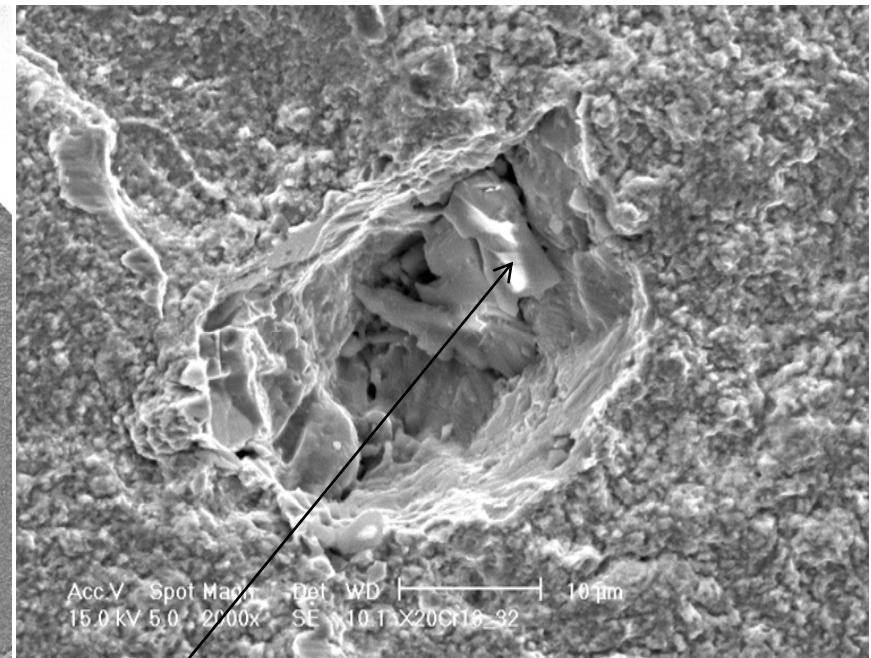
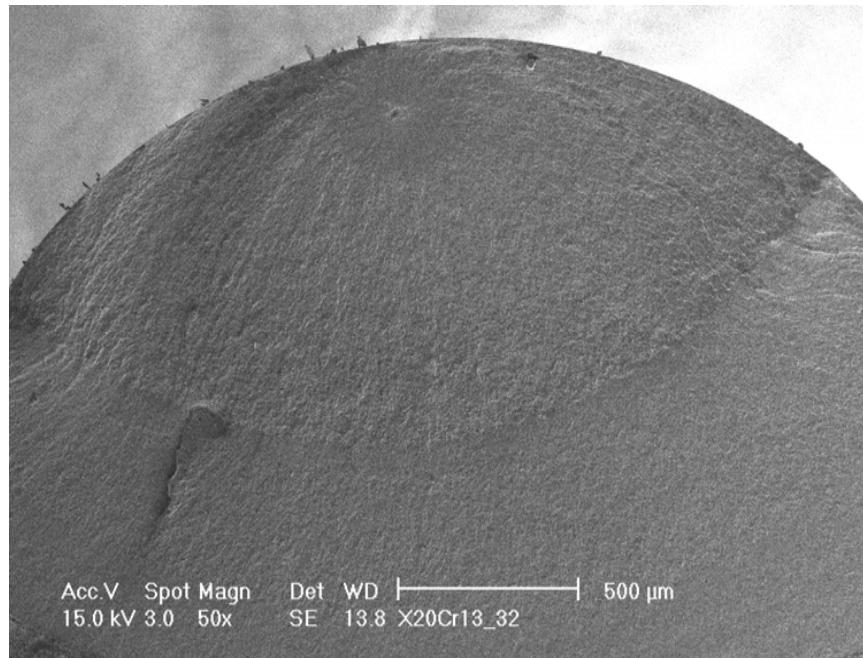
	a)	b)
Al:	22.40	18.44
S:	24.55	27.06
Ca:	17.74	22.59
Mn:	6.96	5.80
Mg	6.07	4.49
Fe:	5.34	8.81

Stanzl-Tschegg, DGM Meeting Berlin, June 2006

# Crack Initiation from Interior Inclusions

Axial 20 kHz Tension-Compression Loading

X20Cr13: 11.3 Cr, 0.2 C, 0.22 Al (940°C / oil quenched, hardening 400°C/2.5 h)



Interior crack initiation from  
Carbide inclusion

EDAX Analysis (wt%) Fe: 64.10  
Cr: 10.01

Stanzl-Tschegg, DGM Meeting Berlin, June 2006

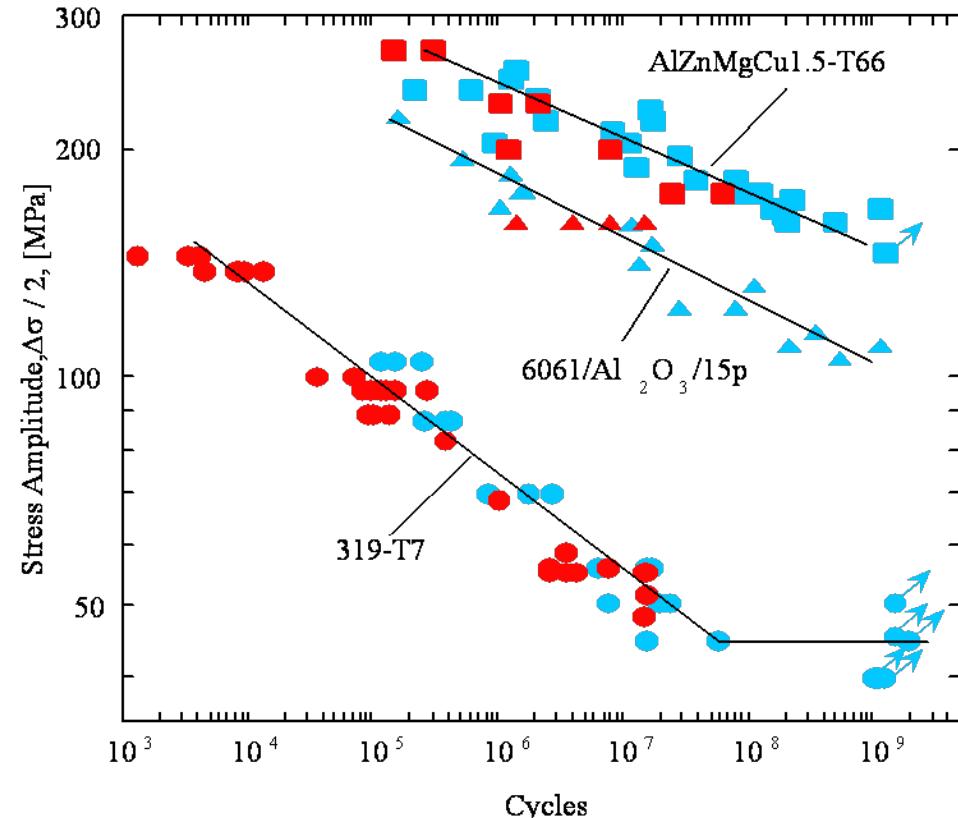
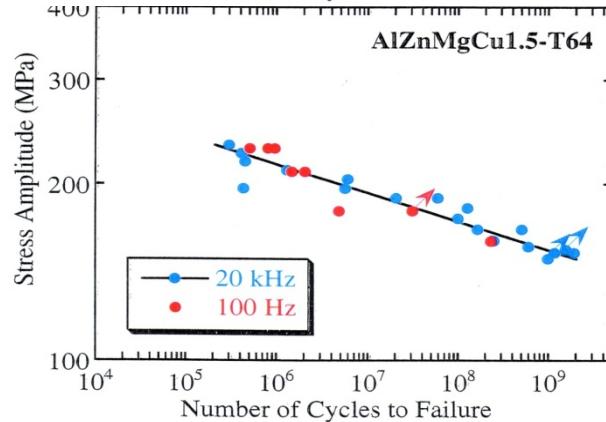
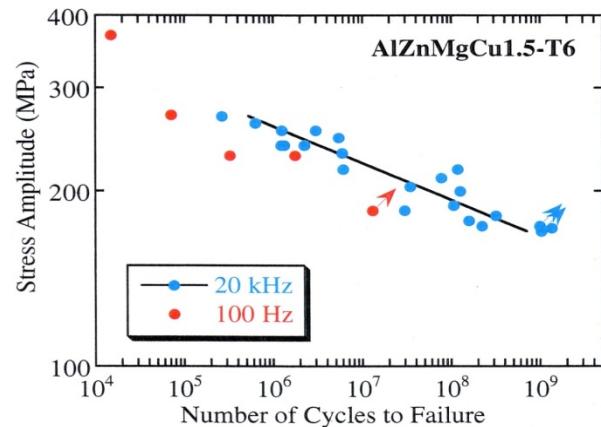


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# Frequency Effect? S-N Curves Al alloys



Mayer H, Papakyriacou M, Pippan R,  
Stanzl-Tschegg SE  
Mat Sci Eng A314, 2001, 48-54

No frequency effect

Papakyriacou M, Mayer H, Fuchs U, Stanzl-Tschegg, Wei RP, FFEMS 25, 2002, 887-896

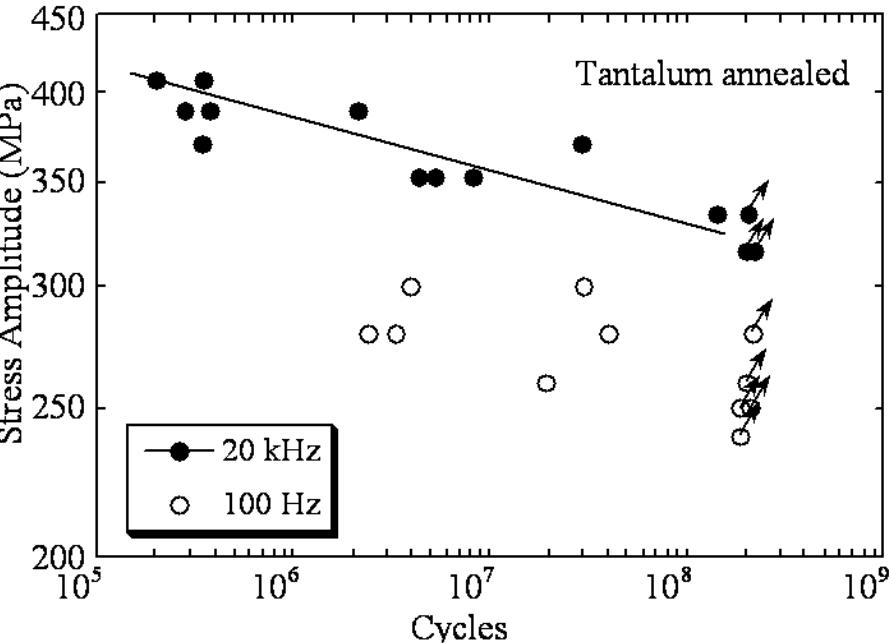
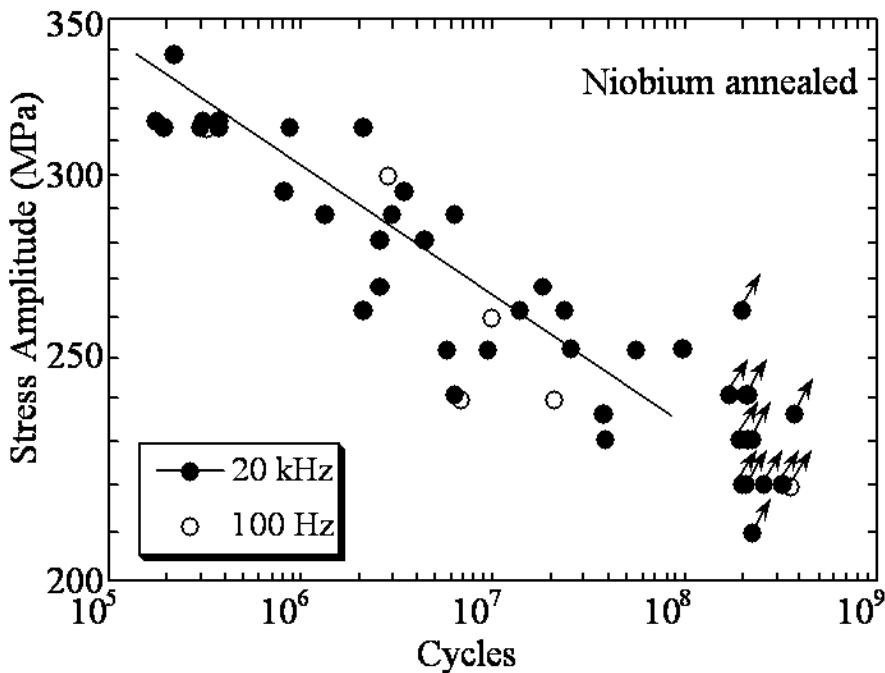


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# Frequency Effect? S-N Curves Nb and Ta



Papakyriacou M, Mayer H, Plenk H, Pypen C, Stanzl-Tschegg SE, Mat Sci Eng A 308, 2001, 143-152

Yield strength  $R_{p0.2} = 385$  MPa  
Endurance limit at  $2 \times 10^8$  cycles:  
240 MPa - 20 kHz    220 MPa - 100 Hz

Niobium: No frequency effect

Yield strength  $R_{p0.2} = 240$  MPa  
Endurance limit at  $2 \times 10^8$  cycles:  
365 MPa - 20 kHz    290 MPa - 100 Hz

Tantalum: Frequency effect

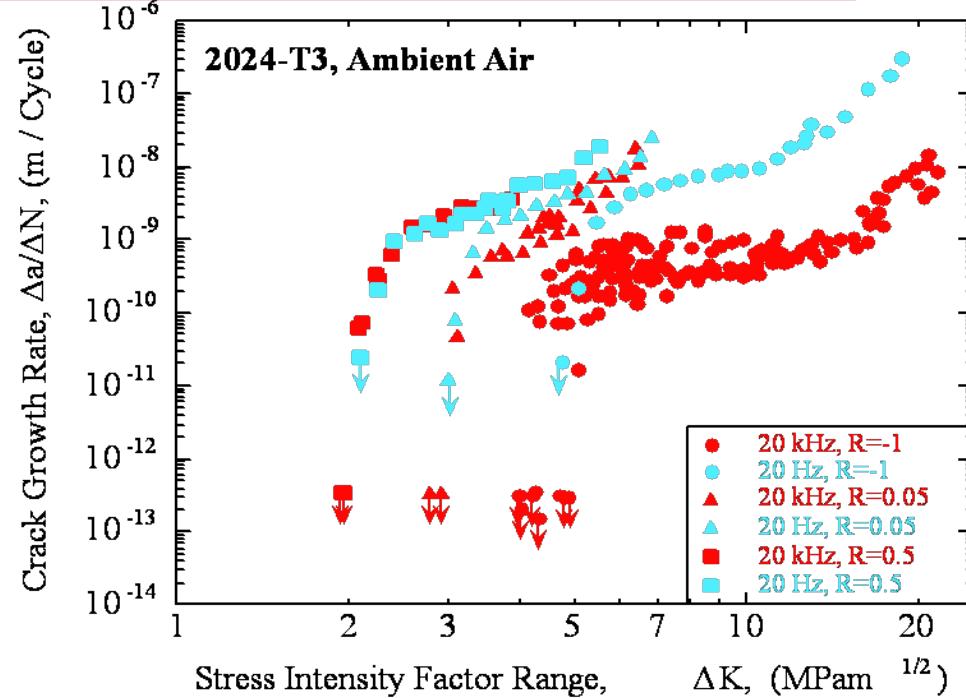
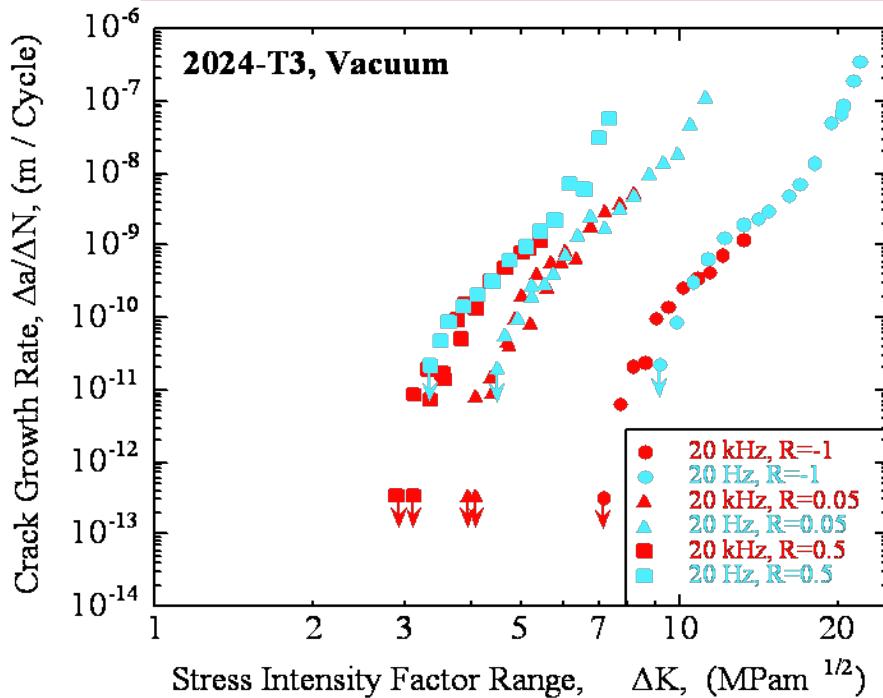


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# Frequency Effect? Fatigue Crack Growth in Humid Air



Holper B, Mayer H, Vasudevan AK, Stanzl-Tschegg, Int J Fatigue, 26, 2004, 27-38

- No influence of plastic deformation rate on  $\Delta a/\Delta N$  (vacuum)
- Frequency effect at  $\Delta a/\Delta N > \text{ca. } 10^{-9}$  m/cycle owing to air humidity
- No frequency effect on  $\Delta K_{\text{threshold}}$



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# Frequency Effect on S-N and $\Delta a/\Delta N$ curves - Summary

Frequency effect present

- At high loads close to yield stress (e.g.Tantalum!)  
→ Testing of strongly damping material problematic
- At higher crack growth rates in corrosive environment

No frequency effect

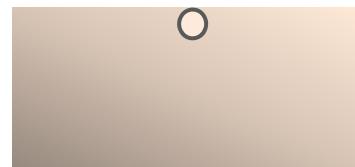
- Al-alloys
- High-strength steels
- Titanium alloys Ti-6Al-7Nb and Ti-6Al-4V
- Mg – high pressure die cast alloy

# Mechanisms of Fatigue Damage

**Crack initiation mostly from  
SURFACE IMPERFECTIONS like**



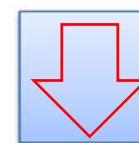
notches,  
scratches



inclusions



PSBs



**or from  
INSIDE**



inclusions,...

**Relevance of PSBs in copper  
for crack initiation, propagation and  
life times (endurance limit?) in the VHCF range**

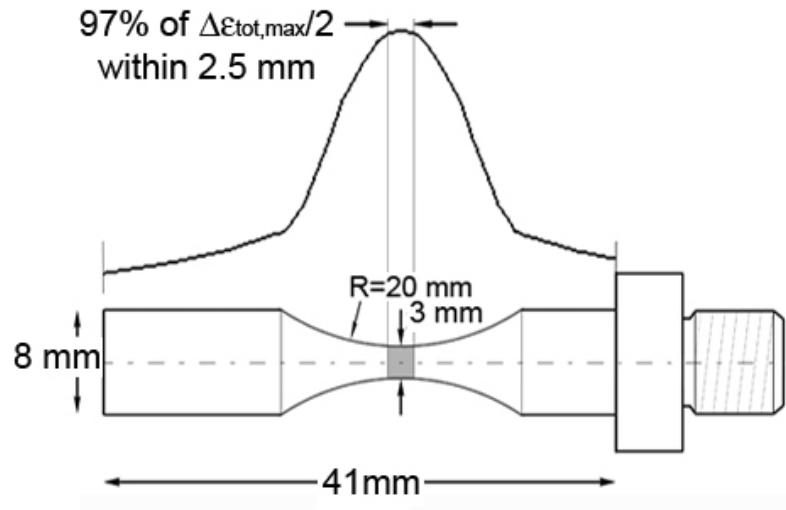


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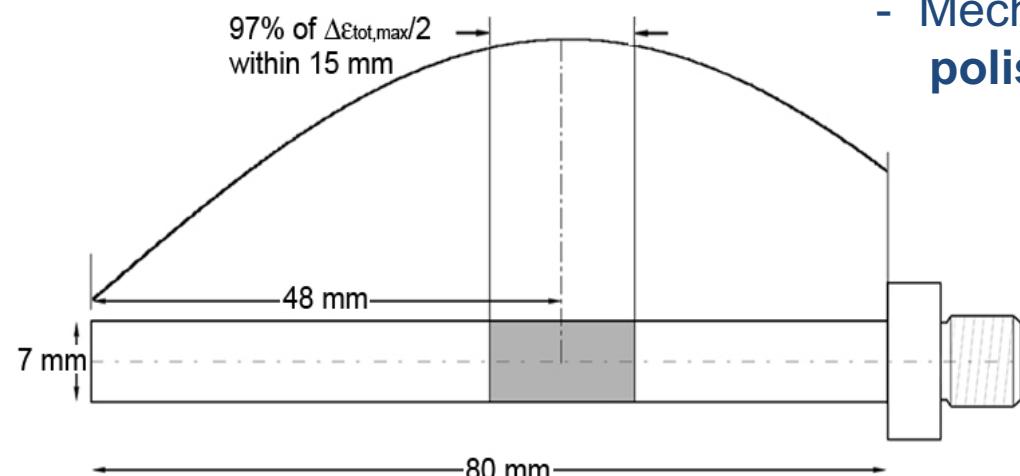
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# Material and Specimen Shape



- Electrolytic copper (DIN1787/17672/1756)
- Cylindrical, 7 mm diameter, 80 mm long Hourglass **shape**, inner diameter: 3mm
- 750°C /75 min, vacuum/ air cooled
- **Mean grain size:** approximately **60  $\mu\text{m}$**
- Mechanical and electrolytical **polishing** (2V, 42 mA/cm<sup>2</sup>)
- **Calibration:**  
Induction sensor at specimen end  
Strain gages along the whole specimen length to measure total axial strain amplitudes  $\varepsilon_{tot}$



# Experimental Procedure

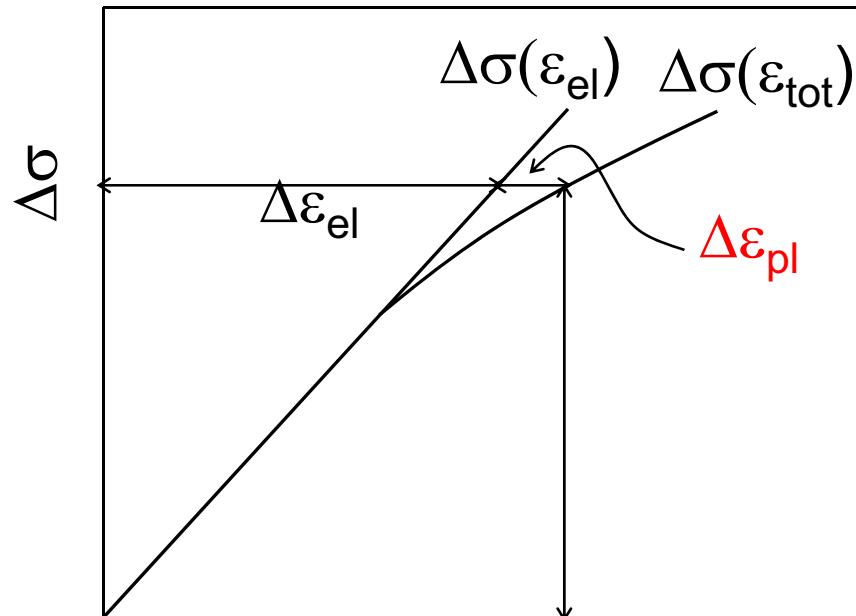
## Ultrasonic test:

- Not stress ( $\Delta\sigma/2$ ) controlled, but
- Displacement or  **$\text{total strain } (\Delta\varepsilon_{\text{tot}}/2)$**   
(strain rate,  $\Delta\varepsilon'_{\text{tot}}/2$ ) **controlled**
- **Plastic strain ( $\Delta\varepsilon_{\text{pl}}/2$ ) control not possible**

*Therefore →*

1. step: Calibration: Determine  $\Delta\varepsilon_{\text{pl}}/2$
2. step: Experiment: Ramp loading

# Determination of Stress Amplitude $\Delta\sigma/2$ in “ductile” material



$\Delta\epsilon_{el}/2$  elastic cyclic strain  
 $\Delta\epsilon_{pl}/2$  plastic cyclic strain  
 $\Delta\epsilon_{tot}/2$  total cyclic strain  
 $\Delta\sigma/2 = \Delta\sigma_{el}/2$  cyclic stress  
 $E = 130 \text{ GPa}$  Young's modulus

Hooke's Law:

$$E = \Delta\sigma / \Delta\epsilon_{el}$$

and with:

$$\Delta\epsilon_{tot} = \Delta\epsilon_{el} + \Delta\epsilon_{pl}$$

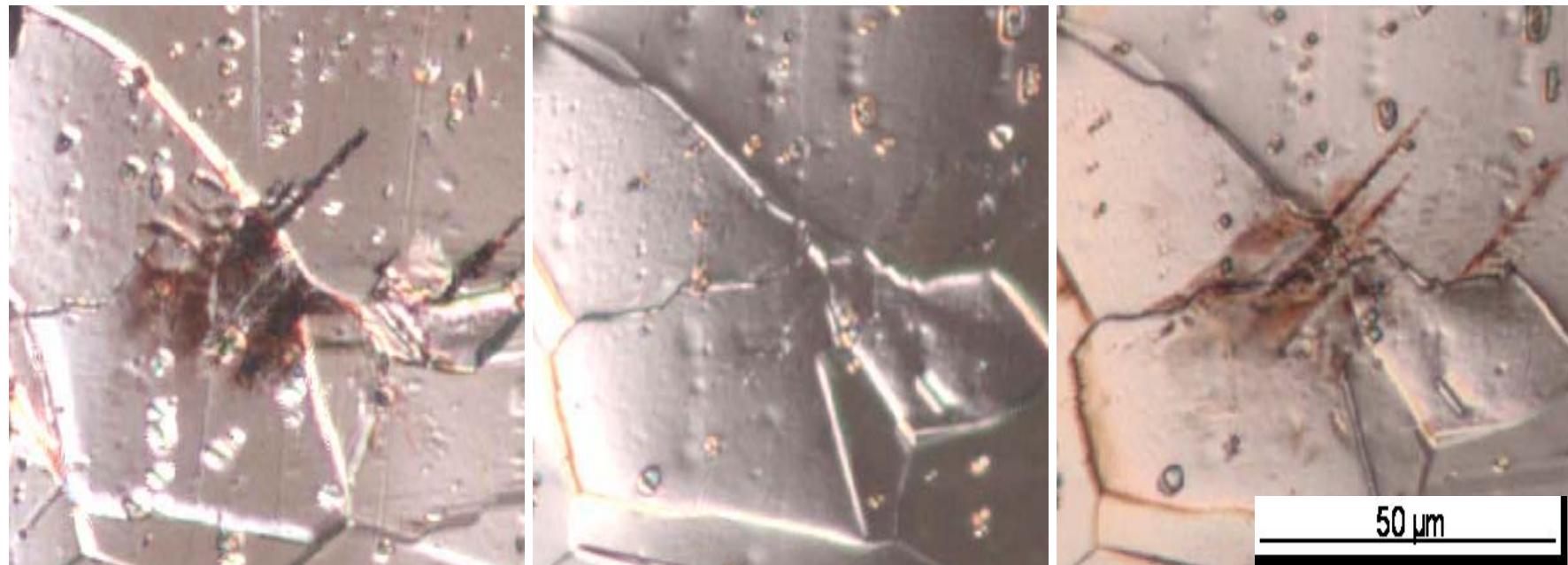
$$\Delta\epsilon_{tot}/2 = \Delta\epsilon_{el}/2 + \Delta\epsilon_{pl}/2$$

$$\begin{aligned}\rightarrow \Delta\sigma/2 &= E \cdot (\Delta\epsilon_{tot}/2 - \Delta\epsilon_{pl}/2) \\ &= E \cdot \Delta\epsilon_{el}/2\end{aligned}$$

# Identification of PSBs: Re-appearance of Slip Bands after Electropolishing PSBs and Reloading

→ „Conventional“ PSB threshold

$$\Delta\sigma/2 \approx 63 \text{ MPa} \quad (\Delta\varepsilon_{pl}/2 \approx 6.1 \times 10^{-6})$$



**Initial loading  $2 \times 10^6$**

Load direction: top to bottom.

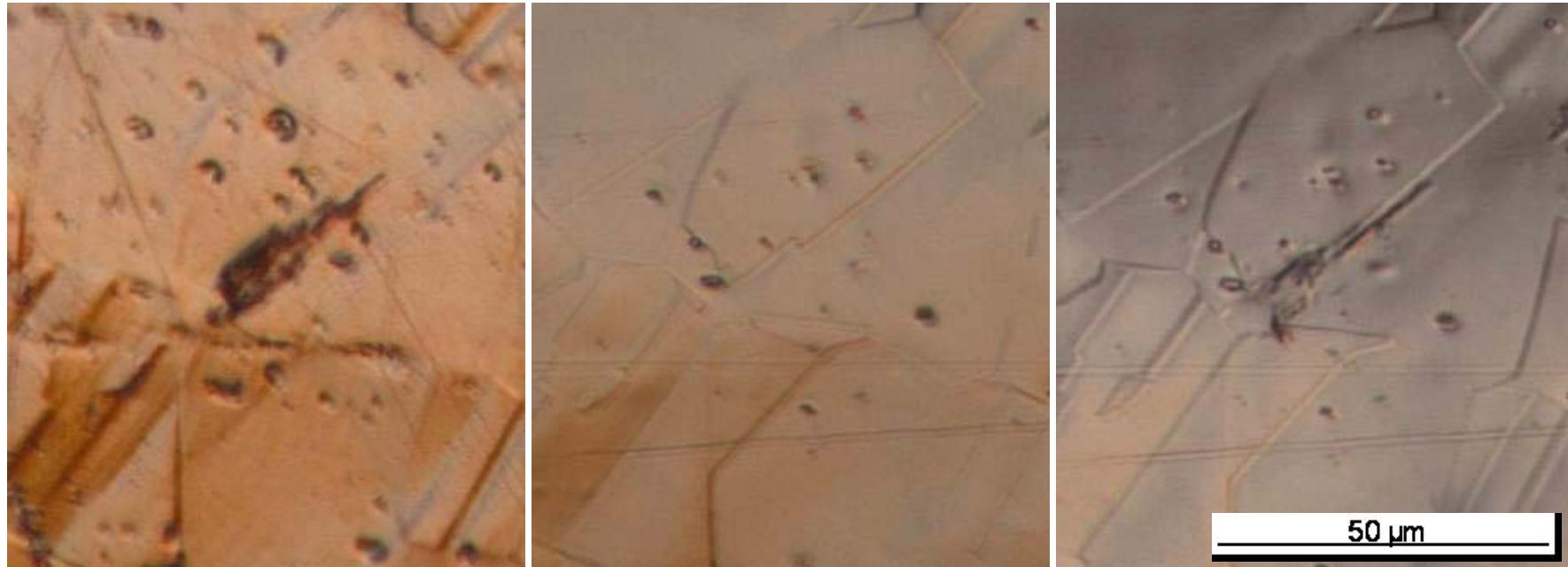
**After polishing**

**+ Reloading  $2 \times 10^6$  cycles**

# PSBs in VHCF range also *below* PSB threshold

→ „VHCF“ PSB threshold

$$\Delta\sigma/2 \approx 45 \text{ MPa} (\Delta\varepsilon_{pl}/2 \approx 3.6 \times 10^{-6}), N \approx 2 \times 10^8$$



$$\Delta\sigma/2 \approx 50 \text{ MPa} (\Delta\varepsilon_{pl}/2 \approx 4.1 \times 10^{-6})$$

Initial loading  $2 \times 10^8$  cycles  
(Load direction: top to bottom)

After polishing

+ Reloading  $2 \times 10^6$  cycles

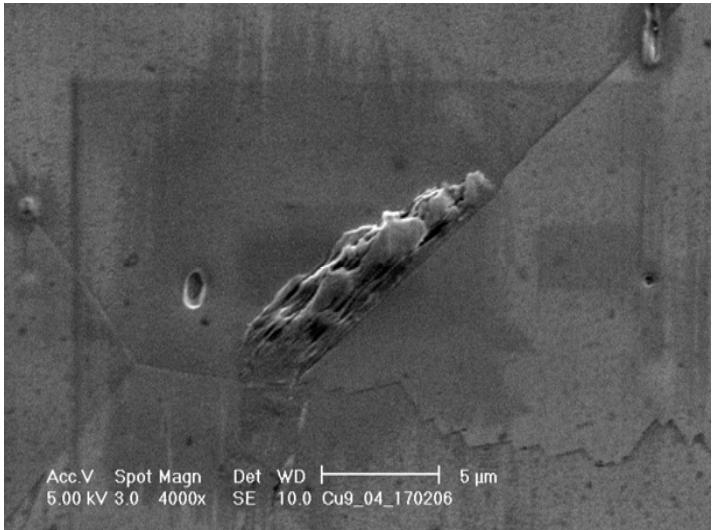


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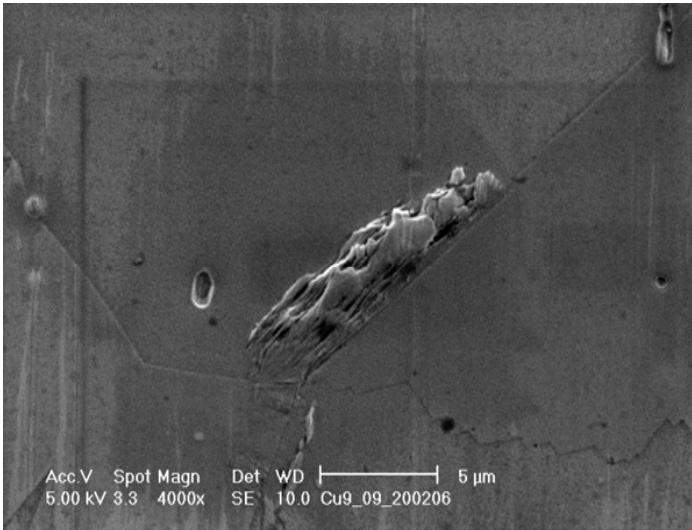
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# Structure and Development of PSBs (SEM)

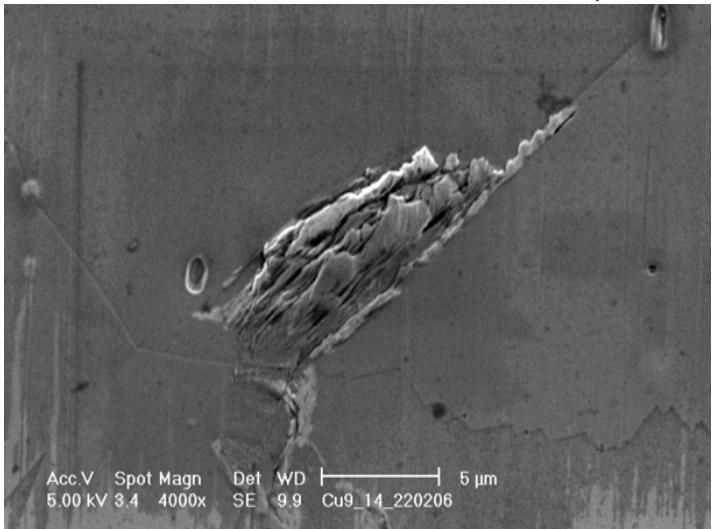


$N = 7,0 \times 10^6$



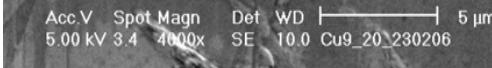
$N = 1,0 \times 10^7$

$\Delta\sigma = 59,2 \text{ MPa}$



$N = 5,0 \times 10^7$

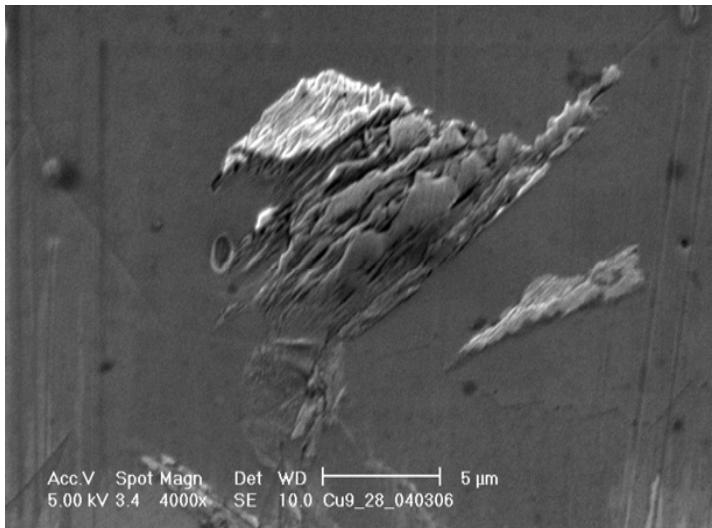
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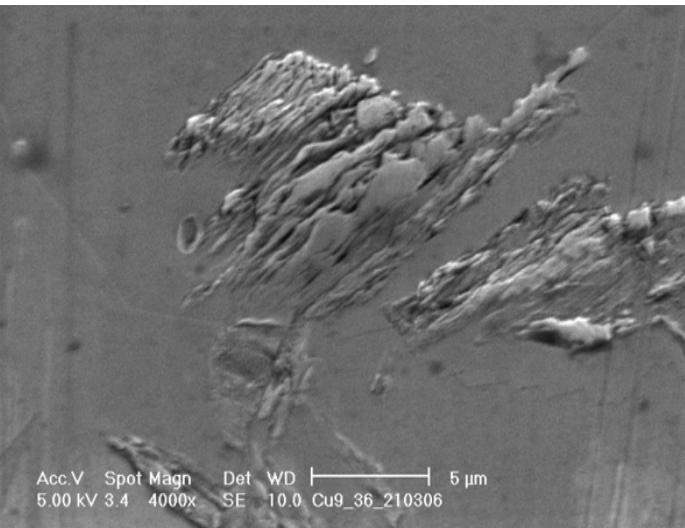
$N = 1,0 \times 10^8$

~ 3.0 MPa  
below  
PSB  
threshold

# Structure and Development of PSBs (SEM)

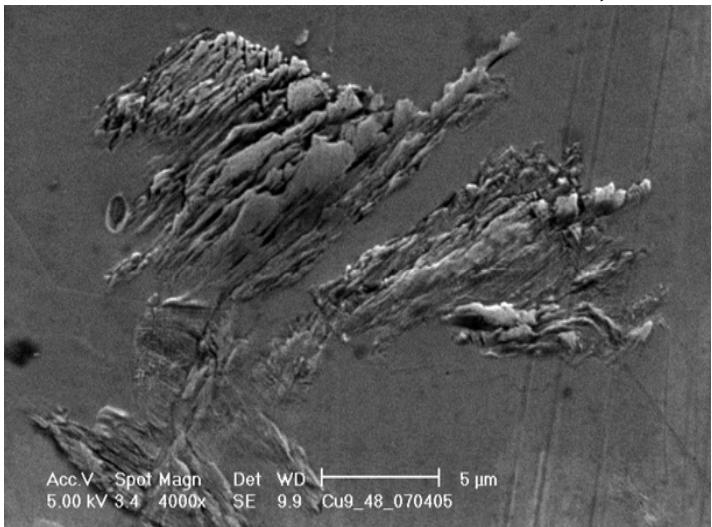


$N = 3,3 \times 10^8$



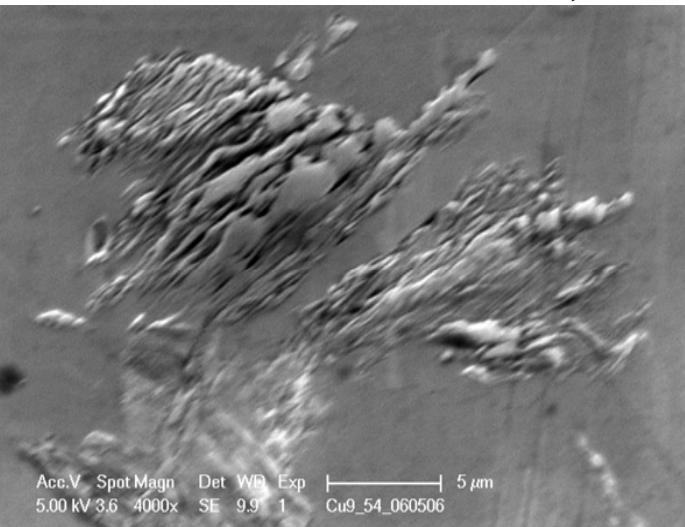
$N = 5,0 \times 10^8$

$\Delta\sigma = 59,2 \text{ MPa}$



$N = 1,0 \times 10^9$

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$N = 2,0 \times 10^9$

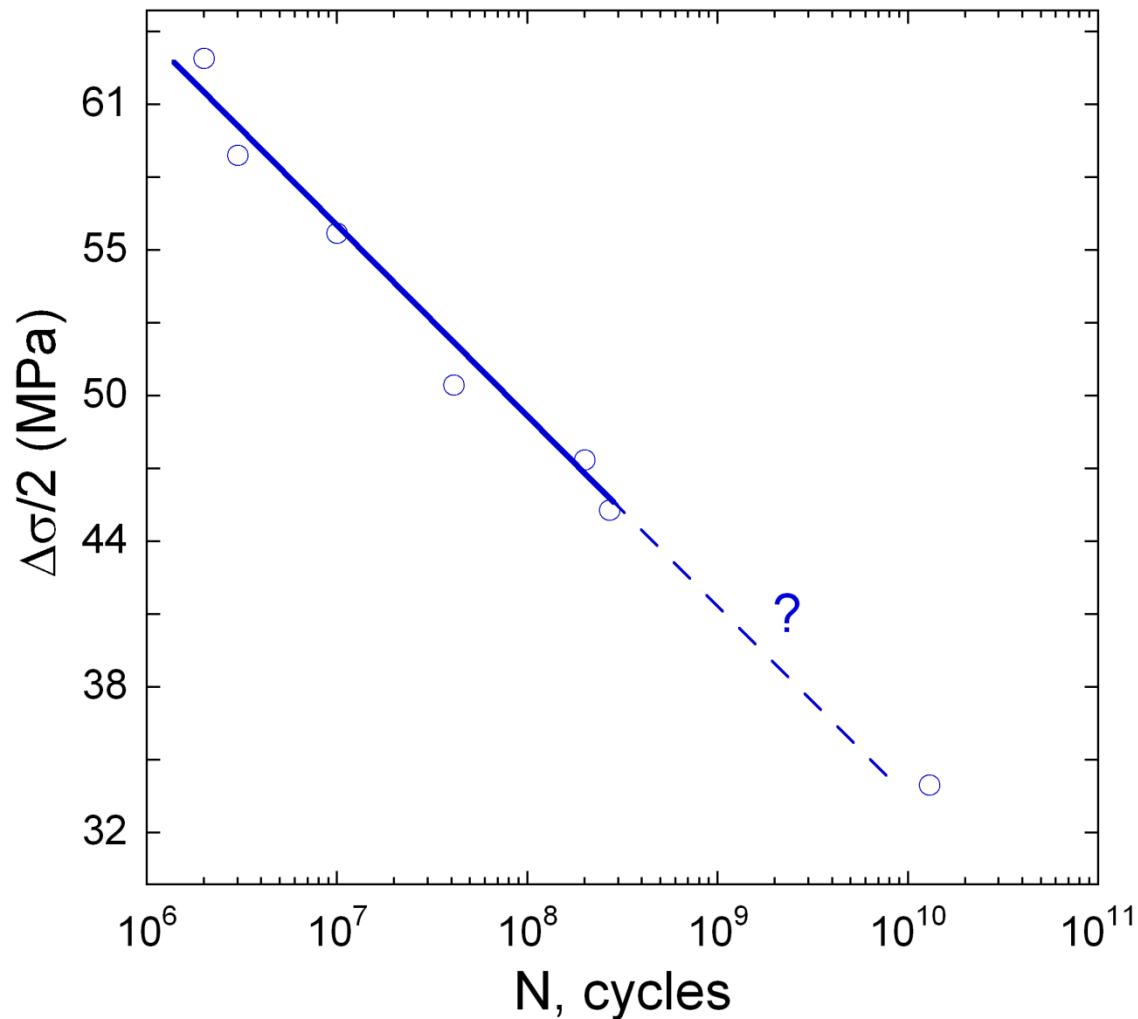
~ 3.0 MPa  
below  
PSB  
threshold



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# Slip Band Formation at and below PSB Threshold: Appearance of first (P)SB

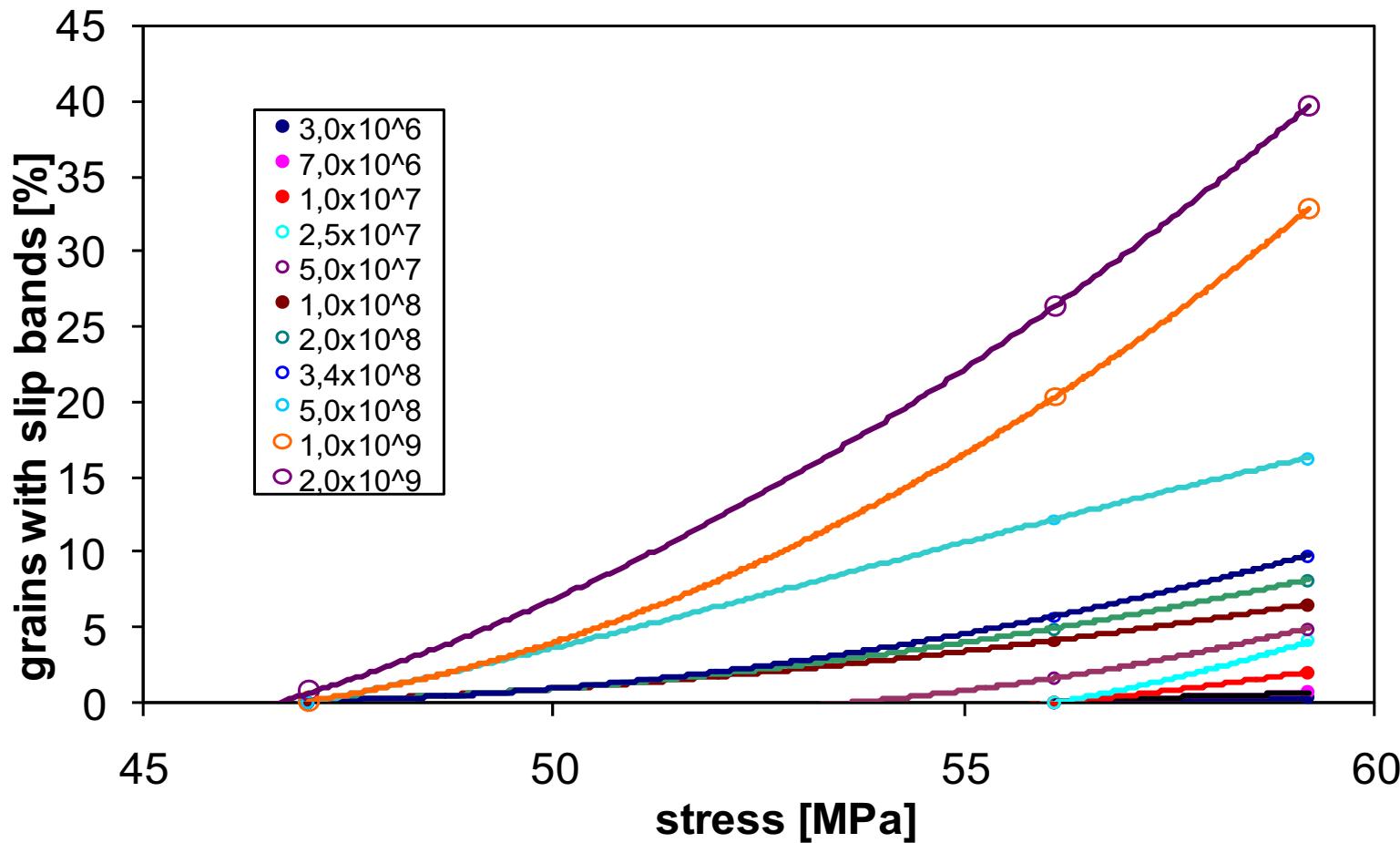


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# Increasing Number of PSBs with $\Delta\sigma/2$ and N



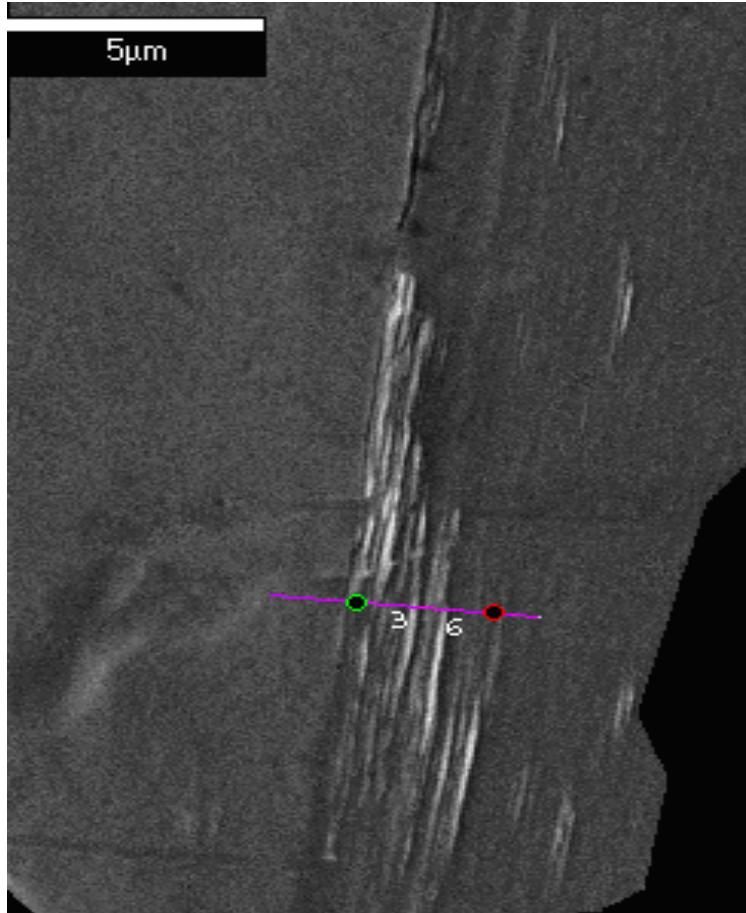
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# PSB Structure and Surface Roughness

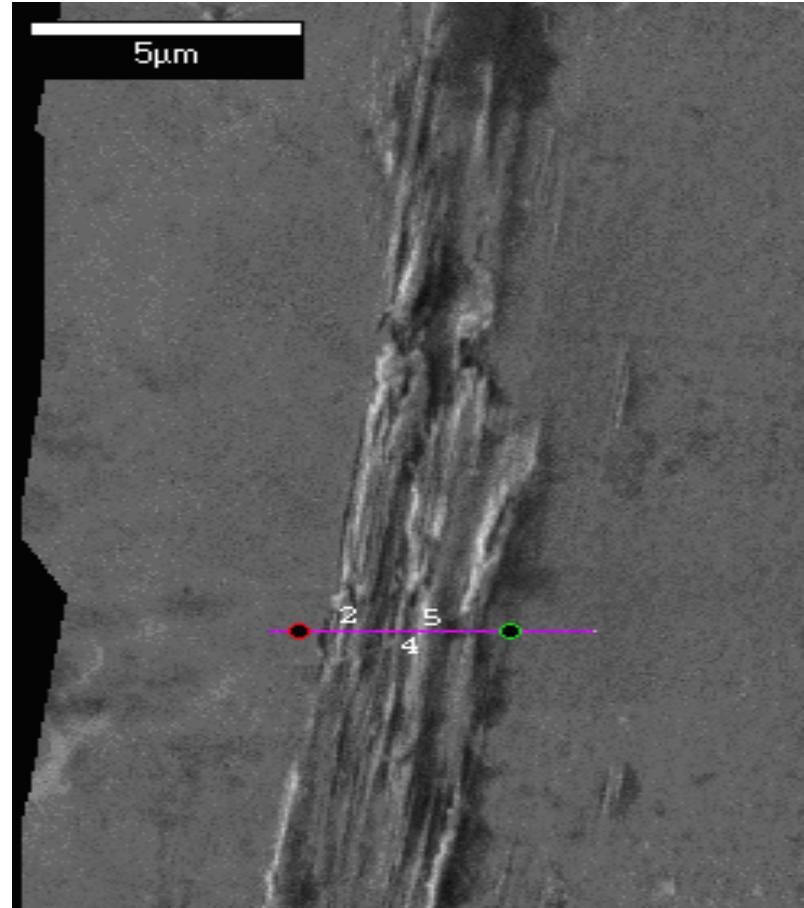
Measurement: **SEM and MeX – 3D** profile analysis software



$\Delta\sigma/2 = 62.1 \text{ MPa}$

$N = 5 \times 10^7 \text{ cycles}$

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$\Delta\sigma/2 = 62.1 \text{ MPa}$

$N = 5 \times 10^8 \text{ cycles}$

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# PSB Structure and Surface Roughness

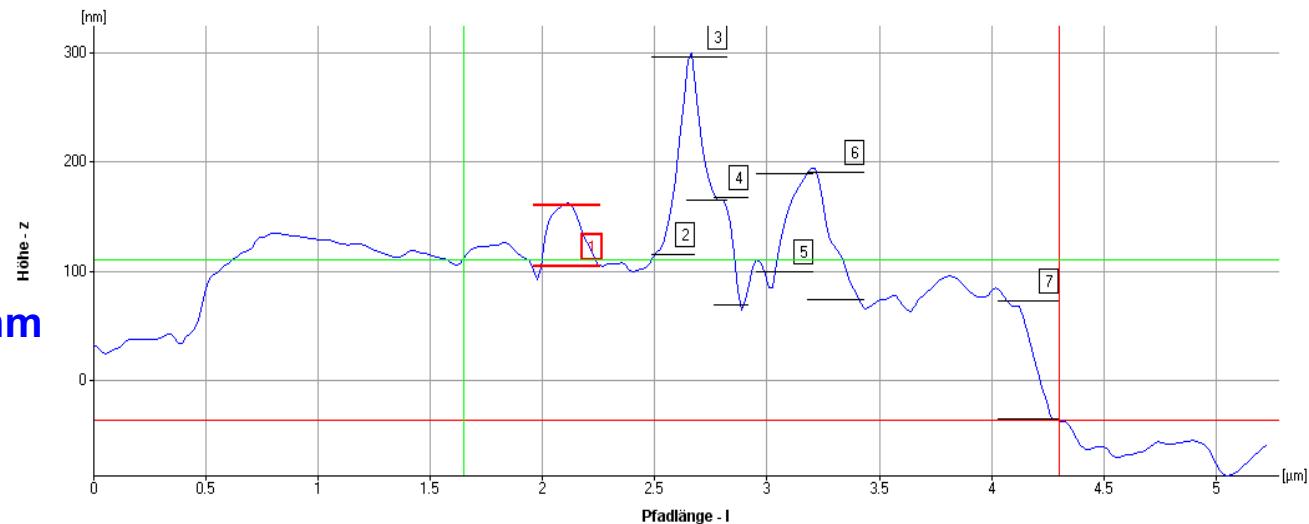
Measurement: **SEM and MeX – 3D** profile analysis software

$$\Delta\sigma/2 = 62.1 \text{ MPa}$$

$$N = 5 \times 10^7 \text{ cycles}$$

mean height = **111,83 nm**

RMS = **117,49 nm**

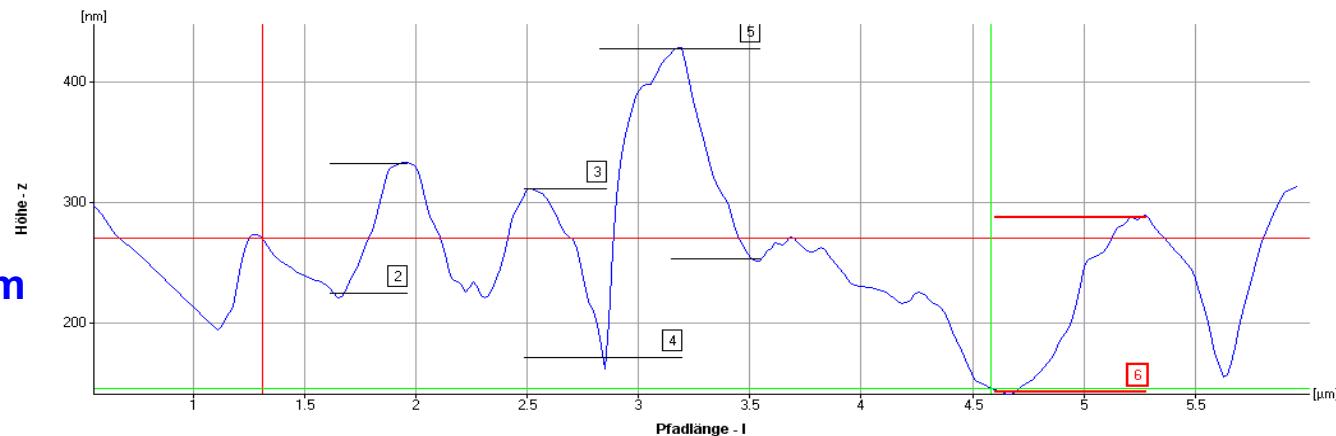


$$\Delta\sigma/2 = 62.1 \text{ MPa}$$

$$N = 5 \times 10^8 \text{ cycles}$$

mean height = **164,8 nm**

RMS = **172,38 nm**



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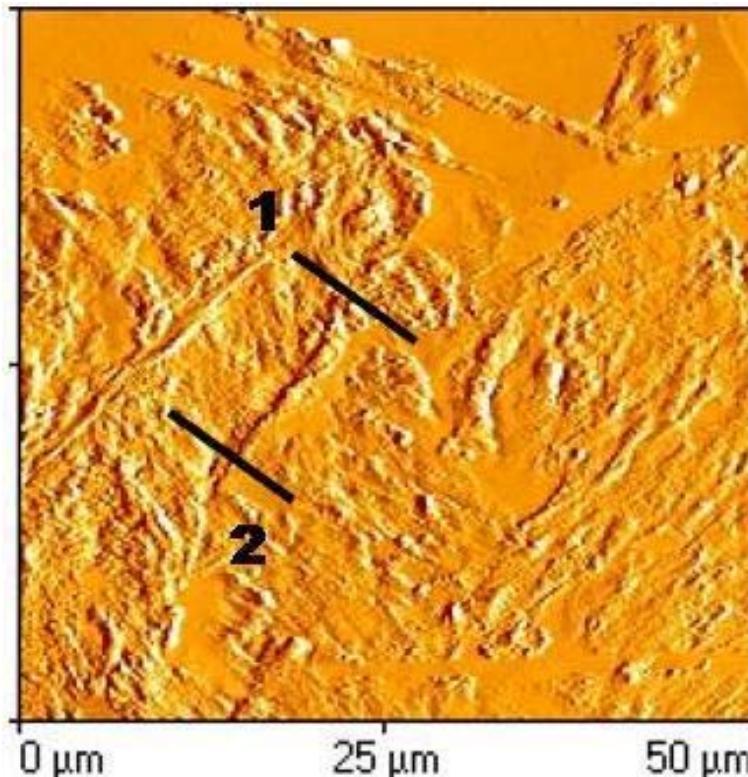
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# Measurement of Surface Roughness

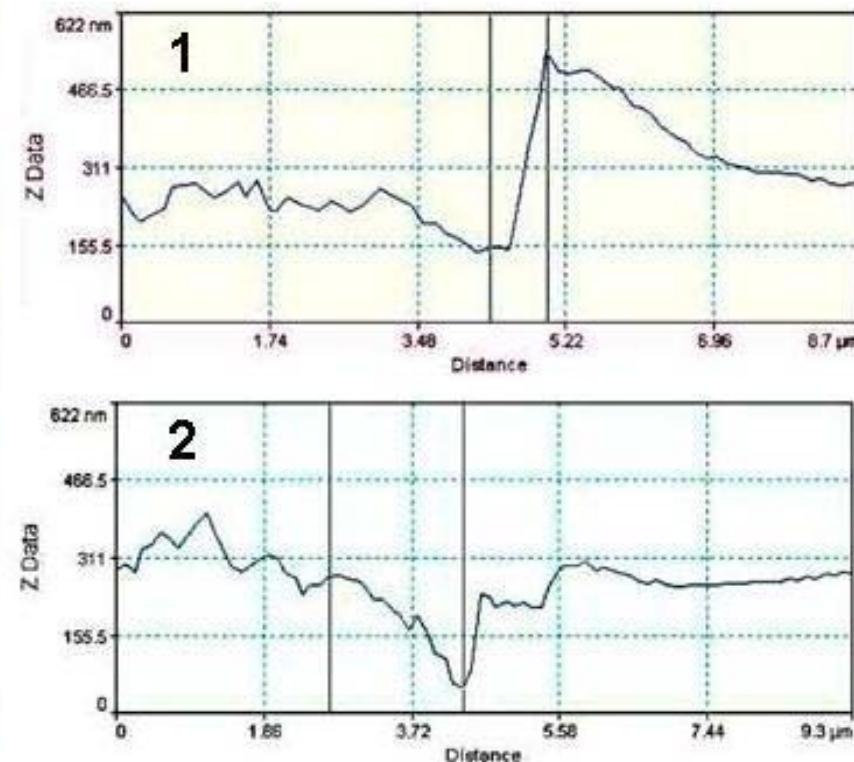
## AFM Topometric Explorer

in contact mode, set point 10 nA, Si<sub>3</sub>N<sub>4</sub> pyramid tip, 50° angle, 50 nm radius



$\Delta\sigma/2 = 53 \text{ MPa}$ ,  $N = 1,28 \times 10^9$  cycles

$h_{\max} \sim 400 \text{ nm}$ ,  $h_{\min} \sim 230 \text{ nm}$



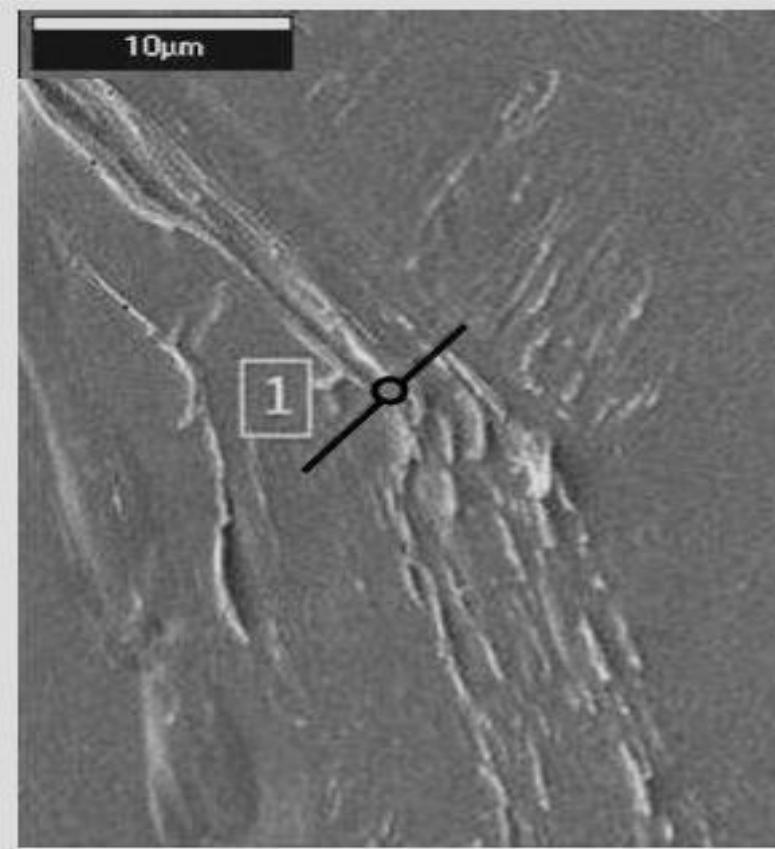
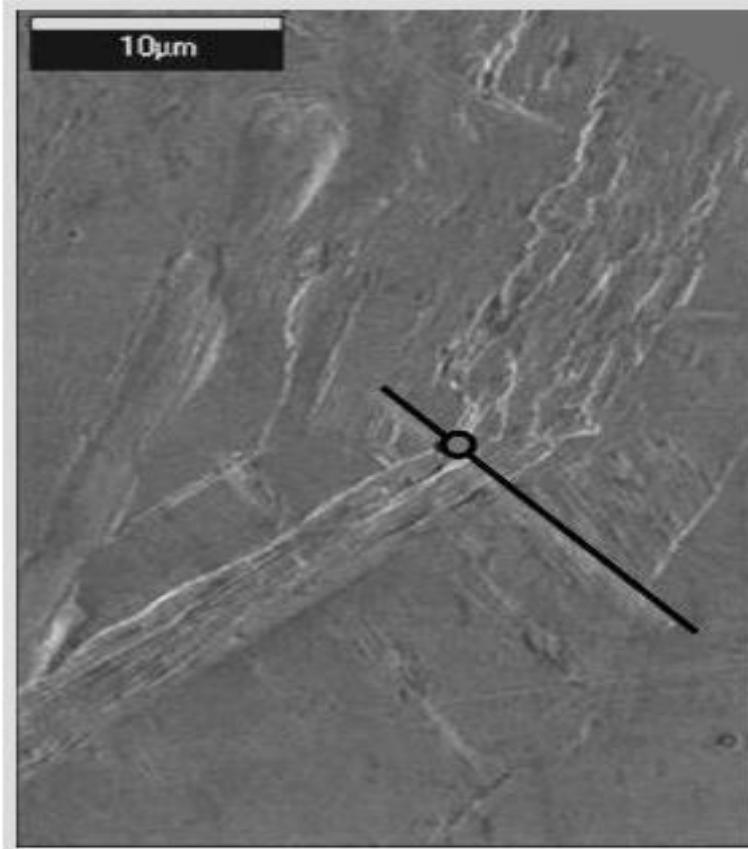
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# PSB Structure and Surface Roughness

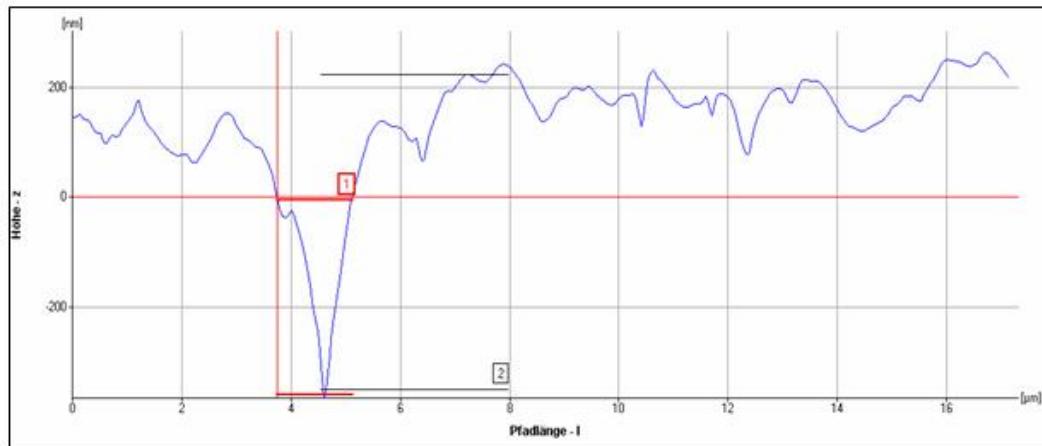
## Plastic replica and SEM



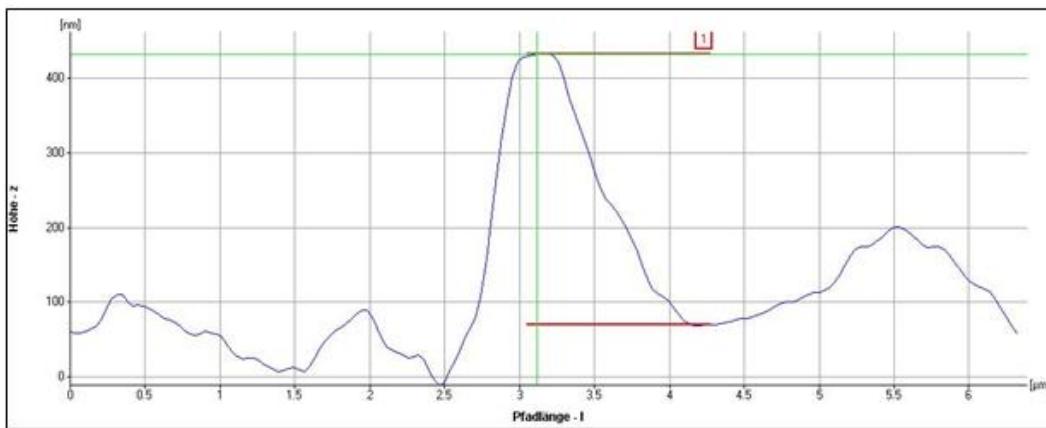
$\Delta\sigma/2 = 100 \text{ MPa}$ ,  $N = 1 \times 10^7$  cycles

# Surface Roughness

MeX – 3D profile analysis software



(a)



(b)

$$\Delta\sigma/2 = 100 \text{ MPa}$$
$$N = 1 \times 10^7$$

Direct SEM:  
Intrusion depth:  
353 nm

Replica:  
Intrusion depth:  
362 nm

# Summary: Surface Roughness

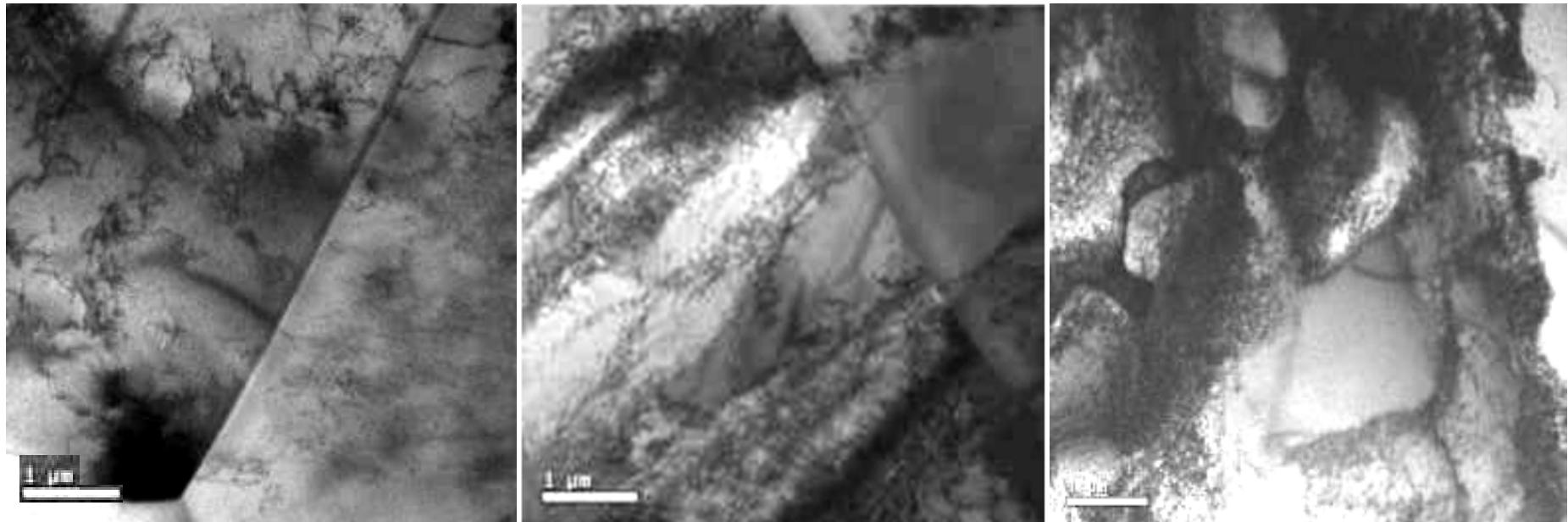
Increase of surface roughness with  $\Delta\sigma/2$  and N

$\Delta\sigma/2$	Roughness			
	Mean	Min.	Max.	
34 MPa	~ 20 nm	~ 14 nm	~ 30 nm	
↓	↓	↓	↓	
60 MPa	~ 200 nm	~ 80 nm	~ 350 nm	
↓	↓	↓	↓	
100 MPa	~ 400 nm	~200 nm	~1000nm	

E.g. Surface roughness ~ 200 nm for  $\Delta\sigma/2 \sim 60$  MPa,  $N \sim 10^{10}$  (mean of three measuring techniques) - high local variation.

# Dislocation structure (TEM)

$\Delta\sigma/2 \approx 64 \text{ MPa}$  ( $\sim 1 \text{ MPa}$  above PSB Threshold ( $\leq 63 \text{ MPa}$ ))  
 $1,5 \times 10^{10} \text{ cycles}$



Dislocations clusters and cells

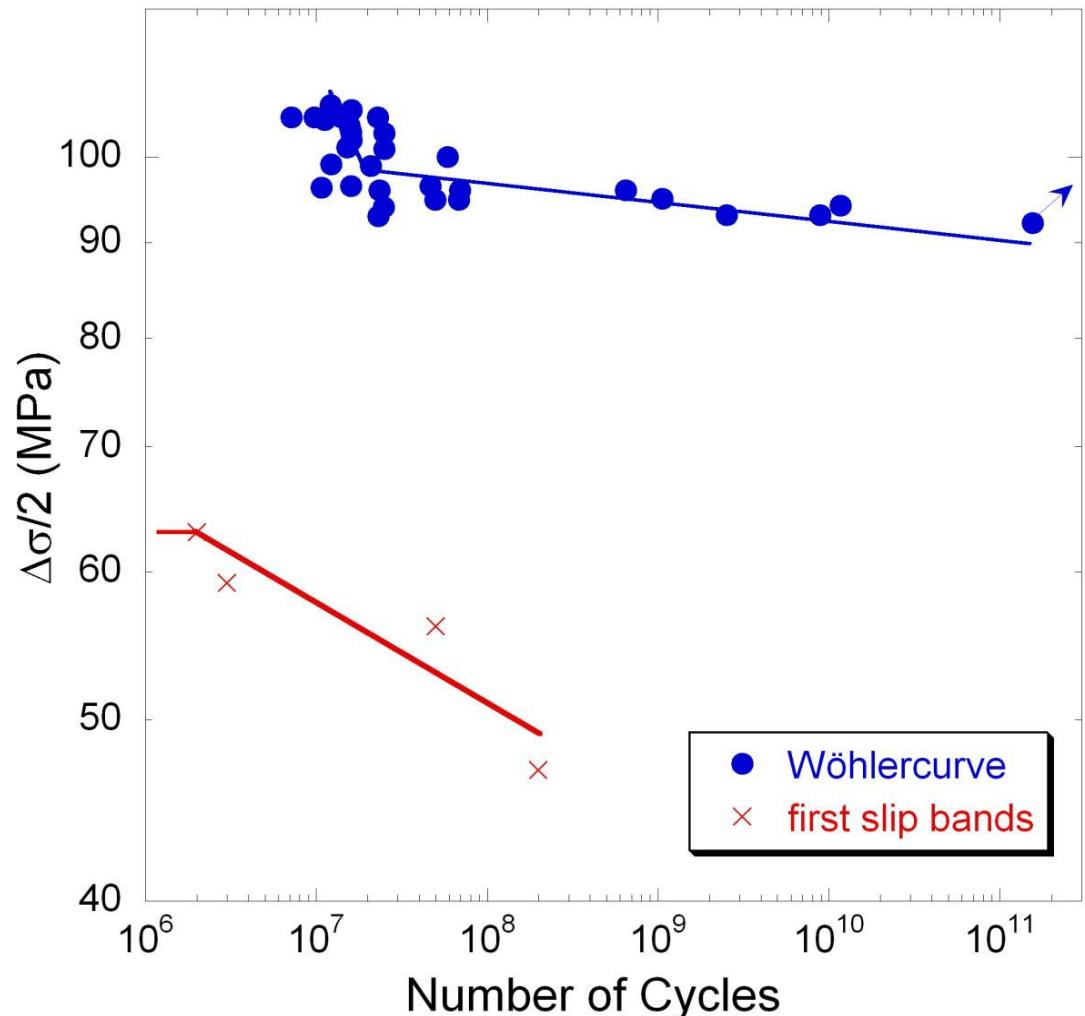


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# Fatigue life diagram, 19 kHz: Wöhler (S-N) plot



**Failure:**

$$\Delta\sigma/2 \sim 93-97 \text{ MPa}$$

$$N_f \sim 3 \times 10^7 - 1.1 \times 10^{10}$$

**No failure:**

$$\Delta\sigma/2 \leq 92.2 \text{ MPa}$$

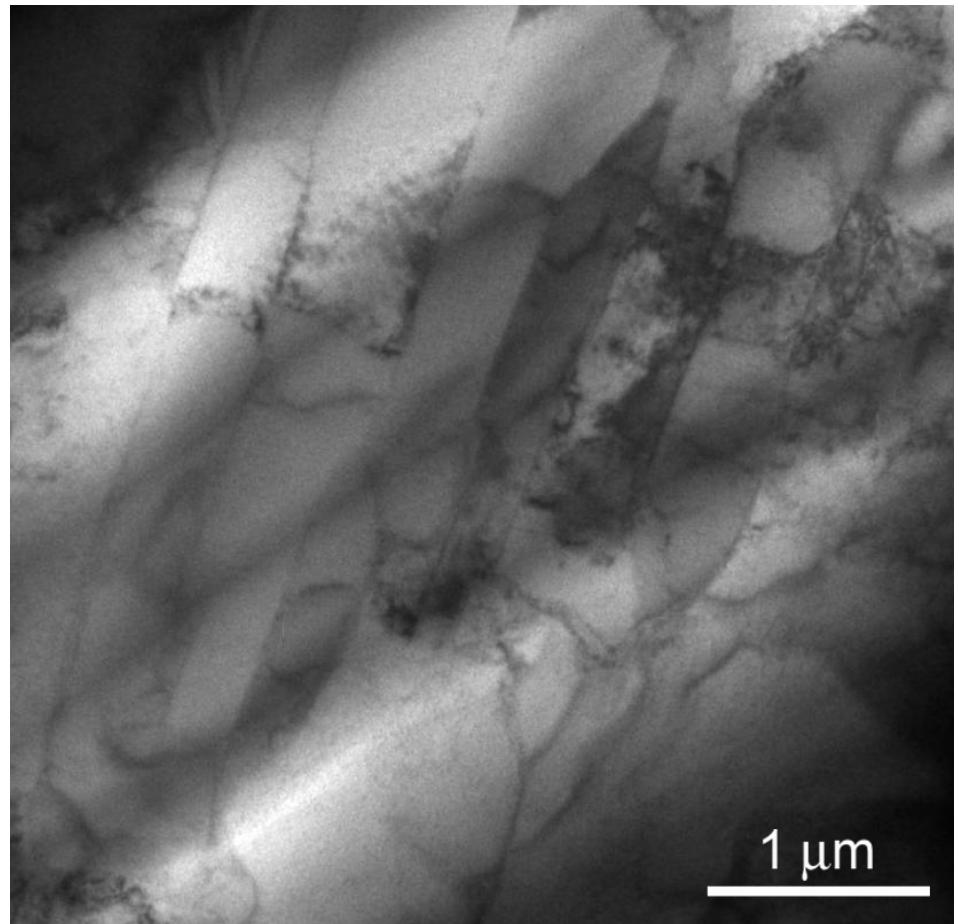
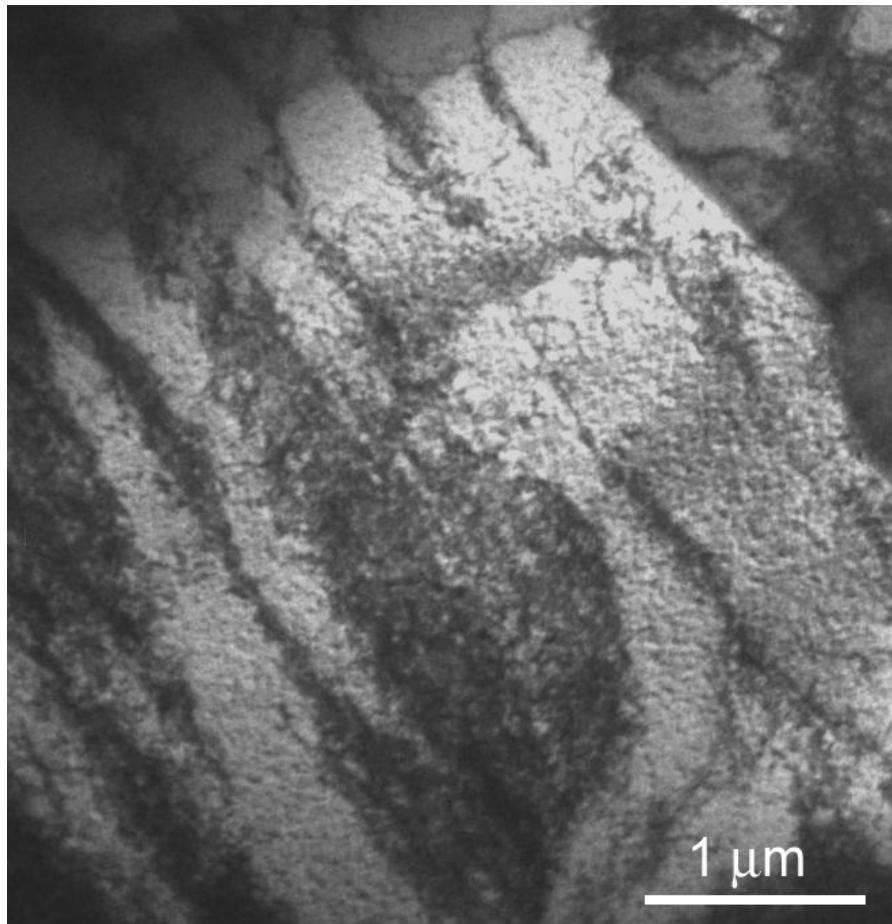
until  $N = 1.7 \times 10^{11}$  cycles

**„VHCF“ PSBs:**

$$\Delta\sigma/2: \sim 45 - 61.7 \text{ MPa}$$

$$N \sim 10^6 - 2 \times 10^8 \text{ cycles}$$

# Dislocation structure (TEM) close to endurance limit



$\Delta\sigma/2 \approx 94 \text{ MPa}$ ,  $N = 1,2 \times 10^{10}$  ~ 1 MPa above fatigue limit ( $\leq 93 \text{ MPa}$ )



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# PSB Threshold (“conventional” as well as “VHCF”) and Endurance Limit are Different

	„Conventional“ PSB Threshold ( $N \sim 10^6$ )	„VHCF“ PSB Threshold ( $N \sim 2.6 \times 10^8$ )	„Endurance Limit“ ( $N_f = 1.14 \times 10^{10}$ )
$\Delta\varepsilon_{tot}/2$	$4.88 \times 10^{-4}$	$3.49 \times 10^{-4}$	$7.27 \times 10^{-4}$
$\Delta\varepsilon_{pl}/2$	$6.1 \times 10^{-6}$	$3.6 \times 10^{-6}$	$2.1 \times 10^{-5}$
$\Delta\sigma/2$	63 MPa	45 MPa	93 MPa



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*How does  $\Delta K_{th}$  correlate with  $\Delta \sigma_o$ ?  
→ Fracture Mechanical Measurements*

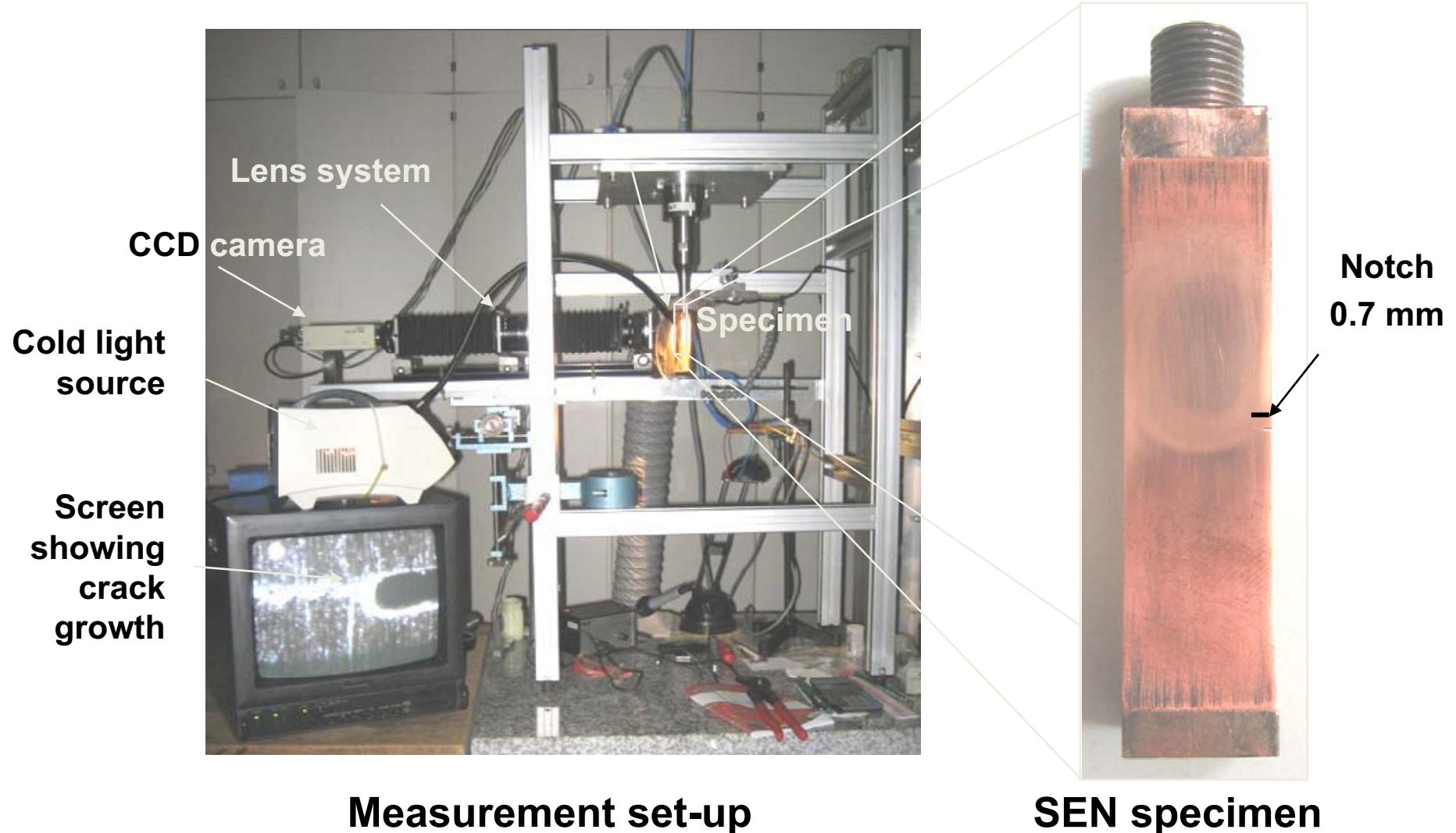


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# Measurement of Crack Propagation



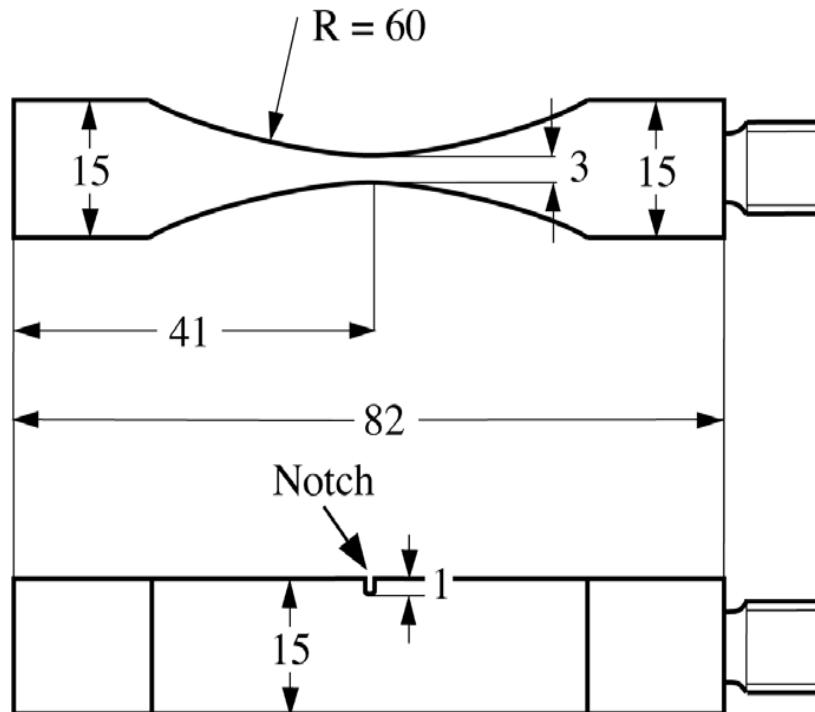
# Ultrasound Fracture Mechanics

Crack propagation and cyclic stress intensities

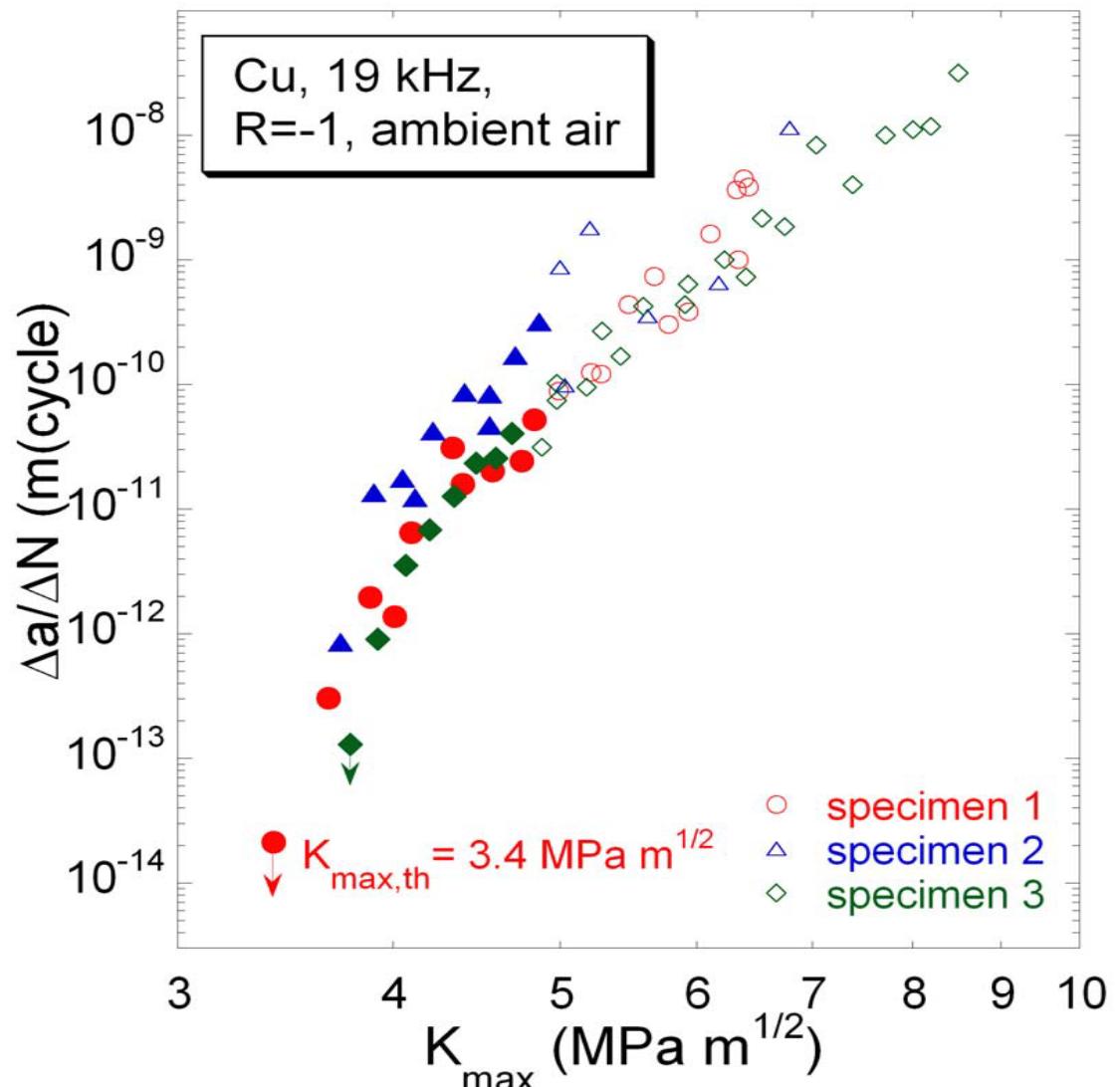
Notched specimens

$$K_{\max} = \epsilon_{\max} \cdot E \cdot \sqrt{\pi a} \cdot Y\left(\frac{a}{W}\right)$$

$$Y = 1.12 - 0.04 \cdot \left(\frac{a}{W}\right) + 0.53 \cdot \left(\frac{a}{W}\right)^2 + 7.8 \cdot \left(\frac{a}{W}\right)^3 - 24.0 \cdot \left(\frac{a}{W}\right)^4$$



# $\Delta a/\Delta N$ vs. $\Delta K/2$ ( $= K_{\max}$ ) Curve

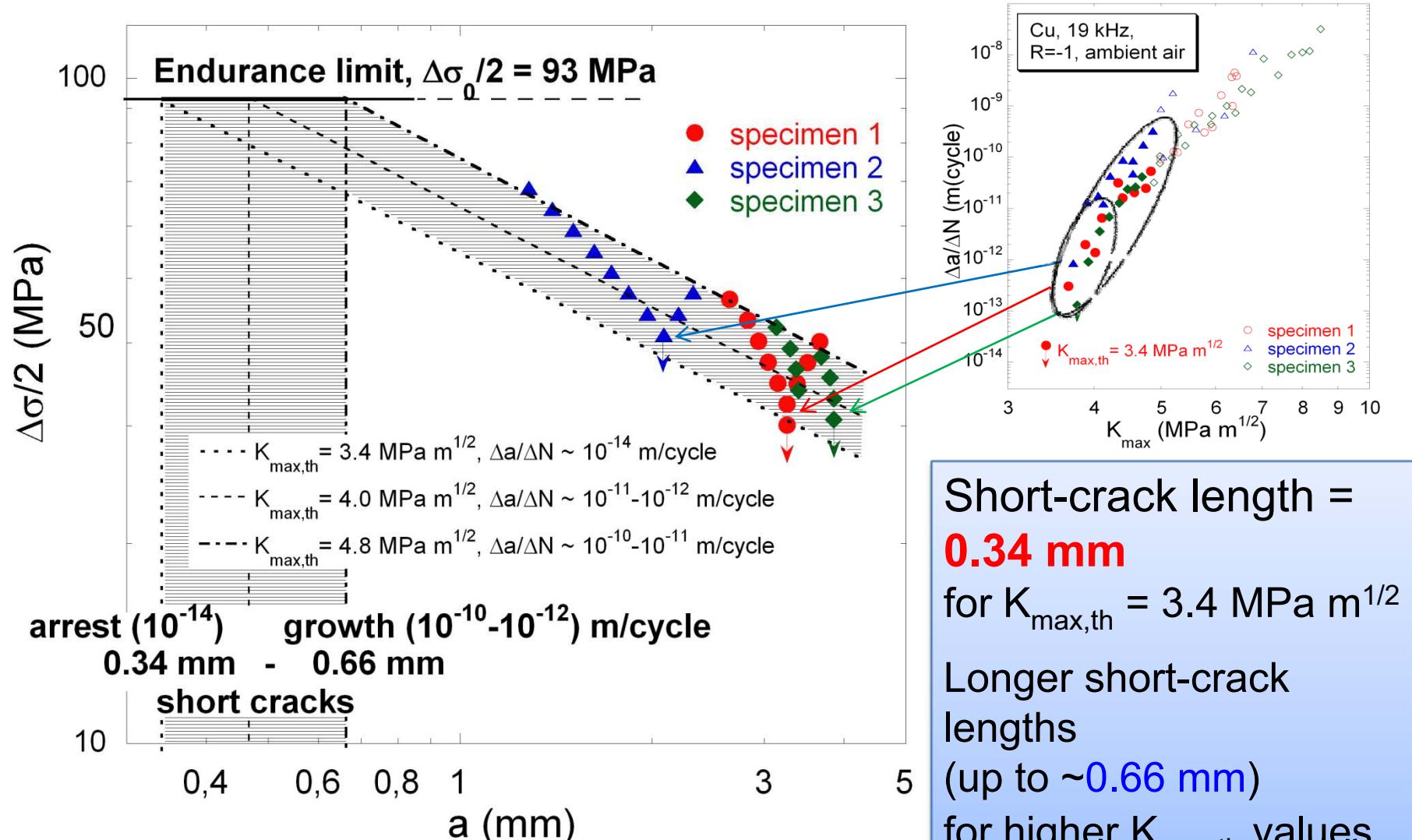


Long crack threshold

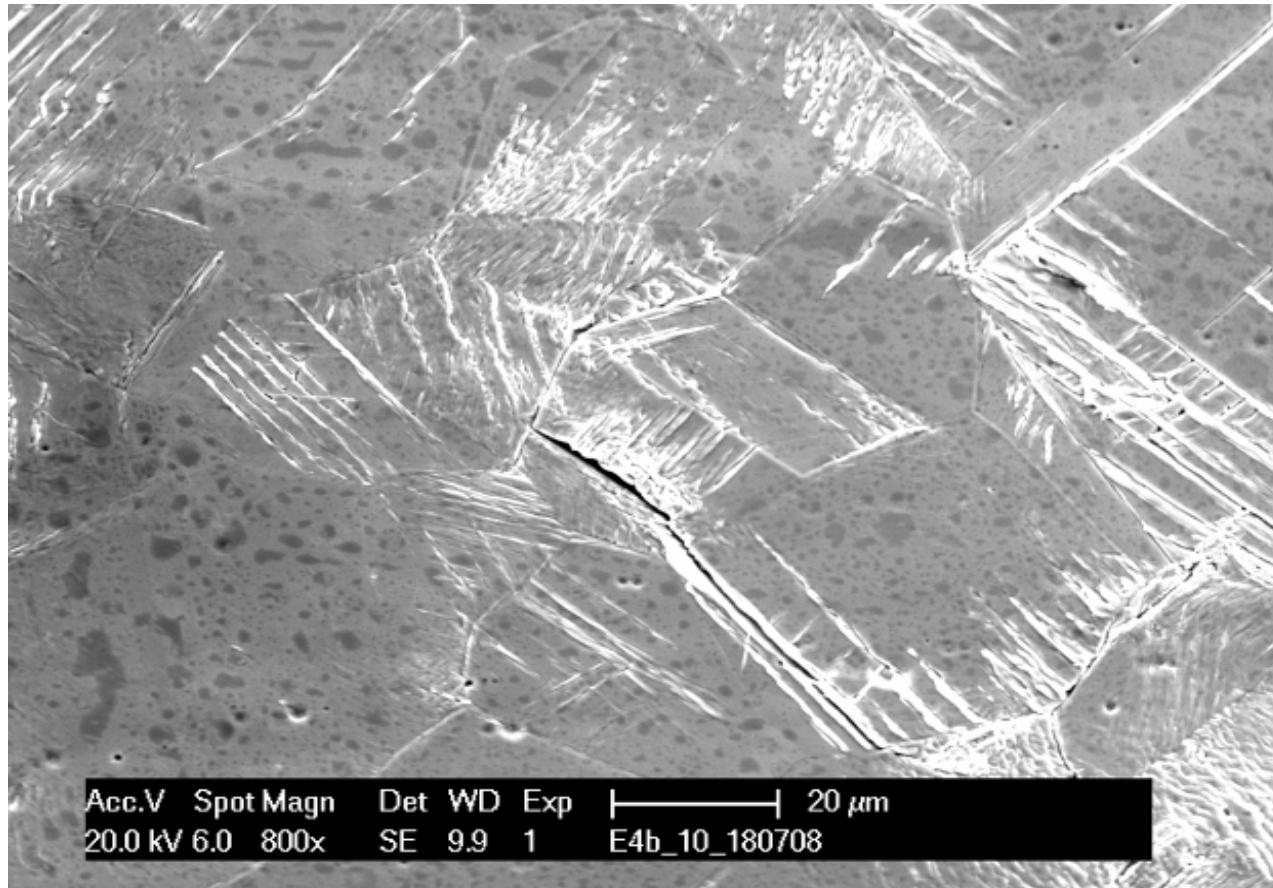
$$K_{\max,th} = \\ 3.4 - 4.8 \text{ MPa } m^{1/2}$$

for  $\Delta a/\Delta N = 10^{-14} - 10^{-10}$   
m/cycle

# Comparison $\Delta K_0$ - $\Delta\sigma_0/2$ (Kitagawa Diagram)



# Reasons for difference of PSB Threshold and Endurance Limit?



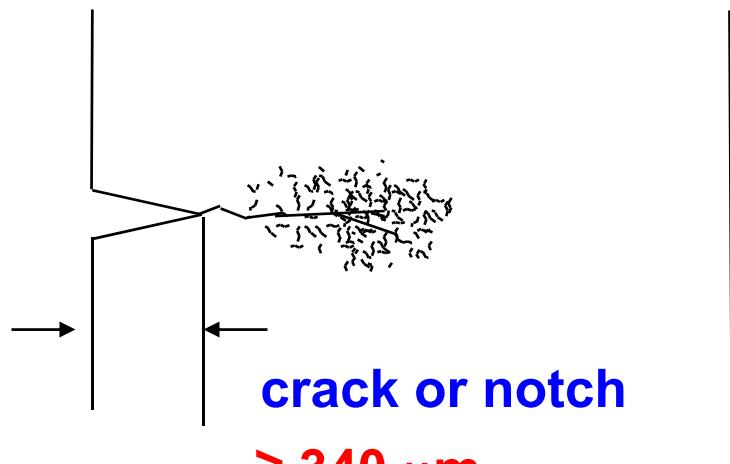
Numerous small cracks, but *no long crack* below endurance limit ( $\Delta\sigma/2 \sim 87$  MPa)

# Formation of LONG cracks above endurance limit, but only SMALL cracks below endurance limit

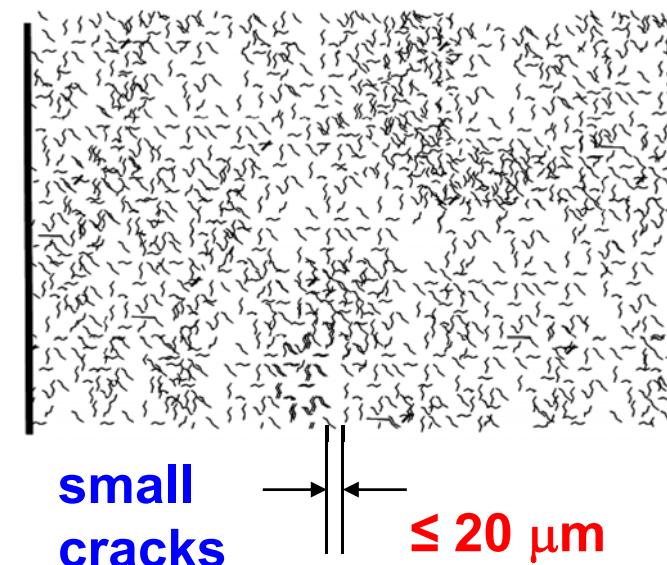
Formation of long propagating cracks (VHCF, 19 kHz)

- at  $\Delta\sigma$  and  $\Delta\varepsilon_{pl}$  values  $\sim 50\%$  above “conventional” PSB threshold
- necessary “short” crack (or notch) length (from Kitagawa diagram)  $\geq 340 \mu\text{m}$
- at “conv.” PSB threshold crack length actually only  $\leq 20 \mu\text{m}$

$\Delta\sigma \geq 93 \text{ MPa}$ :



$\Delta\sigma \leq 63 \text{ MPa}$



# **Evolution of Damage (VHCF, 19 kHz)**



**Below endurance limit**, above PSB threshold:

**numerous non-propagating small cracks**

→ stress relaxation of loaded volume

**Above endurance limit,  $\Delta\sigma$  or  $\Delta K$  high enough for**

**long propagating crack** → **failure**



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# Summary

- **Ultrasonic fatigue technique** most efficient for VHCF testing.
- **S-N curves** beyond  $10^8$  up to several  $10^{11}$  cycles → question on existence or non-existence of **fatigue limit** may be answered („fish-eye“ fractures...).
- Analogous: ( $\Delta a/\Delta N$  vs.  $\Delta K$ ) **curves** down to  $10^{-14}$  m/cycle → (non)-existence of  $\Delta K_{\text{threshold}}$
- **No frequency effect** for most technical alloys.
- **Frequency influence** for very ductile materials and in corrosive environment ABOVE  $\Delta K_{\text{threshold}}$

# Summary

- Tests on **polycrystalline copper**:
- PSBs form below „conventional“ PSB threshold (63 MPa)  
**PSB formation depends on amplitude AND number of cycles** („VHCF threshold“: 45 MPa)
- **Endurance limit** ( $\Delta\sigma_0$ ): Failure at  $\Delta\sigma = 93$  MPa after  $1.1 \times 10^{10}$  cycles. No failure below  $\Delta\sigma_0$  within  $2 \times 10^{11}$  cycles.
- **Correlation of  $\Delta K_{th}$ ,  $\Delta\sigma_0$  and relevance of PSB threshold**  
Kitagawa diagram →  
Critical short-crack length is  $\sim 340$   $\mu\text{m}$  for  $\Delta K_{th}$   
**At/below PSB threshold:** actual crack length  $\sim 20\mu\text{m}$  →  
No long crack propagation → **no failure**



*Thanks for  
your  
attention!*



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5 novembre 2008*



*Stefanie Stanzl-Tschegg*

