Effect of yttrium content on the ultra-high cycle fatigue behavior of Mg-Zn-Y-Zr alloys

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Abstract In the super-long life regime, the fatigue behavior of as-extruded Mg-6wt%Zn-xY-0.8wt%Zr Mg alloys with Y content of 0, 1, 2 and 3 wt% have been investigated, respectively. The result indicates that for all measured S-N curves, a plateau exists in the regime of 5×10^{6} - 10^{8} cyc, and then the fatigue strength gradually decreases between 10^{8} and 10^{9} cyc. Therefore, only fatigue strength corresponding to 10^{9} cyc can be determined. Compared with other alloys, the alloy with Y content of 2 wt% has the highest fatigue strength and its value is 105 MPa. SEM observations to fracture surfaces reveal that for all alloys, the fatigue crack mostly initiates at the surface or subsurface of samples failed within 10^{6} - 10^{9} cyc. Further observation indicates that the crack initiation is related with activated slip bands instead of phase particles and activated twins. Based on the measured results and Murakami equation, it demonstrates that the fatigue strength of alloys is more dependent on the hardness values.

Keywords Mg alloy, super-long fatigue, slip band, fatigue properties

1. Introduction

Magnesium alloys are currently used in cars for low stress applications such as covers and less frequently for the mechanically loaded structural components such as wheels, transmission housings and pedals [1]. Generally, car wheels need to be stressed at different amplitudes for several 10^8 cycles in service. For casting alloys, defects such as casting porosity and cavities are usually present and the fatigue properties are affected significantly by their shape and dimension [1-2]. Several material defects such as casting porosity, oxidation films and intermetallic inclusions, can act as crack initiation sites and reduce material's fatigue strength in the super-long fatigue life regime [1-2]. Therefore, the fatigue strength of as-cast Mg alloys corresponding to 10^9 cycles is generally about 40-50 MPa [1]. In contrast, wrought alloys are basically defect-free and have superior mechanical properties, thus the evaluation of their fatigue properties is of great interest for understanding the intrinsic fatigue mechanism of Mg alloys [3]. In early reported work, researchers mainly focused on the fatigue behavior of wrought Mg alloys with the fatigue lifetime less than 10^7 cyc [2, 4, 5]. However, as for the ultra-high cycle $(10^7 - 10^9 \text{ cyc})$ fatigue behavior of wrought Mg alloys, only a few research papers can be referred [6-8]. Additionally, when compared with other system Mg alloys, Mg-Zn-Y-Zr alloys are much stronger [9-10], and their tensile strength can reach up to 380 MPa. Previous work demonstrated that Y content can remarkably influence the microstructure and tensile properties of Mg-Zn-Y-Zr system alloys [9-12]. Therefore, it can be predicted that the change of Y content should have some effect on the fatigue behaviour of Mg-Zn-Y-Zr system alloys. However, so far, no related literature can be referred. The aim of this work is to disclose the crack initiation mechanism and to establish the relationship between Y content and the fatigue strength in the gigacycle regime by investigating the fatigue behavior of Mg-5.65%Zn-xY-0.8%Zr alloys with Y content of 0, 1, 2 and 3 wt%, respectively.

2. Experimental procedure

The materials used in this study were the as-extruded Mg-Zn-Y-Zr alloys with different Y contents, which were prepared by special technology in magnesium alloy research department of

IMR, China. Through inductively coupled plasma atomic emission spectrum (ICP-AES) apparatus, the chemical compositions of alloys I-IV were determined, as listed in Table 1. The extrusion ratio was 10:1. Vickers hardness testing was performed with a load of 250g (HV).

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Nominal	Composition (wt %)					цV
alloy	Mg	Zn	Y	Zr	Zn/Y	11 v
Alloy I	Bulk	5.68	0	0.78		62 ± 5
Alloy II	Bulk	5.53	1.08	0.83	5.12	78 ± 4
Alloy III	Bulk	5.64	1.97	0.73	2.86	74 ± 4
Alloy IV	Bulk	5.49	3.08	0.82	1.78	68 ± 4

Table 1 Chemical composition of the as-extruded Mg-Zn-Y-Zr alloys

The specimens used for the super-long fatigue study were taken along extrusion direction (ED) of the plates. The dimensions of the specimens for fatigue test are shown in Fig. 1. Fatigue test was conducted on a Shimadazu USF-2000 ultrasonic fatigue testing machine at a resonance frequency of 20,000 Hz, with a resonance interval of 150 ms and a stress ratio of R = -1 in ambient air (temperature of 25-35 °C, relative humidity of 40-60 %). Specimens were cycled at constant amplitude until failure or until at least 10⁹ cycles were reached. After test, fracture surfaces of failed samples were examined using Environmental Scanning Electron Microscope (XL30-FEG-ESEM).



Fig. 1. Dimensions of the ultrasonic fatigue specimen.

3. Results and discussion

Fig. 2 shows the fatigue strength (S) versus lifetime (N) curves of four Mg-Zn-Y-Zr alloys. It can be seen that for all measured S-N curves, a plateau exists in the regime of 5×10^{6} - 10^{8} cyc, and then the fatigue strength gradually decreases between 10^{8} and 10^{9} cyc. Therefore, only fatigue strength corresponding to 10^{9} cyc can be determined. Compared with other alloys, the alloy with Y content of 1 wt% has the highest fatigue strength and its value is 105 MPa.

To understand and compare the crack initiation mechanism of different alloys that failed in the super-long fatigue regime, four representative samples designated as S1, S2, S3 and S4 are chosen for analysis, as shown in Fig. 2. Fig. 3 shows the overall fracture surfaces of samples. It can be seen that for all samples, the fatigue crack preferentially initiates at the surface or subsurface.

Based on the backscattered electron image, the overall fracture surface can be divided into three regions, i.e. crack initiation region (Region 1), steady crack propagation region (Region 2) and tearing region (Region 3), as shown in Fig. 4. It is very interesting to find that basically no phase

particles exist in Region 1, whereas phase particles can be easily observed in Regions 2 and 3. Moreover, the number density of existing phase particles in Region 3 is obviously higher than that in Region 2. Therefore, it suggests that the existence of phase particles can not cause the fatigue crack initiation, but can act as the preferential route for crack propagation. Fig. 5 shows the optical microstructure of the areas just underneath the fracture surface of the sample S3. It reveals that only a few twins can be activated in the Region 1, whereas the activated twins can be easily observed in Regions 2 and 3. Additionally, density of the activated twins in Region 3 is remarkably higher than that in Region 2. In the same way, it indirectly demonstrates that the crack initiation is can not be caused by the activated twins under a stress amplitude far less than the yield stress. After crack initiation, the propagated crack will gradually increase the stress intensity factor near the crack tip, leading to the local plastic deformation and the activation of twins.

Previous work indicated that the micro cracks mainly initiate along the intense slip bands in the grain interior [12], as shown in Fig. 6. Therefore, it firmly demonstrates that the localized slip bands activated at elastic stress amplitude is the main reason for the crack initiation.



Fig. 2. S-N curves of the as-extruded Mg-Zn-Y-Zr alloys with Y content of: a) 0wt%, b) 1wt%, c) 2wt% and d) 3wt%.

Certainly, if the sample surface contains some scratches, it will further increase the occurrence possibility of crack initiation. Generally, the relationship between the "crack initiation area" size and fatigue strength or fatigue limit (σ_{-1}) proposed by Murakami can also be used for Mg alloys, which can be expressed as [13]:

$$\sigma_{-1} = \beta \frac{(\text{HV}+120)}{(\sqrt{A_{\text{in}}})^{1/6}}$$
(1)

Where HV is the Vicks Hardness of Mg matrix, β is a coefficient and A_{in} is the crack initiation area.



Fig. 3. Secondary electron images showing the overall fracture surfaces of samples: a) S1 (95MPa for 5E8 cyc), b) S2 (110MPa for 5.2E8 cyc), c) S3 (105MPa for 5.1E8 cyc) and d) S4 (100MPa for 5.3E8 cyc)

From Fig. 3, it can be seen that the crack initiation area for all failed samples is very similar. Then, the fatigue strength of alloys should be more dependent on the hardness values. Based on the results in Table 1, the hardness of the alloy with Y content of 1wt% is relatively higher. According to Equation (1), the calculated fatigue strength corresponding to 10^9 cycles should be higher than that of others, which is consistent with the measured results (Fig. 2).



Fig. 4. Backscattered electron image showing the phase particle distribution on the fracture surface (S3).



Fig. 5. Optical images of the as-polished fracture surface of sample S3 taken from regions 1-3 in Fig. 4.



Fig. 6. Micro crack initiation at slip bands in the grain interior. [12]

4. Conclusions

For all measured S-N curves, a plateau exists in the regime of 5×10^6 - 10^8 cyc, and then the fatigue strength gradually decreases between 10^8 and 10^9 cyc. Therefore, only fatigue strength corresponding to 10^9 cyc can be determined. Compared with other alloys, the alloy with Y content of 2 wt% has the highest fatigue strength and its value is 105 MPa. Additionally, for all alloys, the fatigue crack mostly initiates at the surface or subsurface in the super-long fatigue lifetime regime.

Acknowledgements

This work was supported by National Natural Science Foundation of China projects under Grant No. 51171192 and No. 51271183, a National Basic Research Program of China (973 Program) project under Grant No. 2012CB067425 and an Innovation Fund of Institute of Metal Research (IMR), Chinese Academy of Sciences (CAS).

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